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2 **An Implantable, Compatible and Networkable**  
3 **Nanocomposite Piezoresistive Sensor for *in situ***  
4 **Acquisition of Dynamic Responses of CFRPs**

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24 submitted to *Composites Science and Technology*  
25 (initially submitted on 29 Oct 2020; revised and re-submitted on 24 Jan 2021)

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27 **Abstract**

28 When sensors are embedded in composites for structural integrity monitoring (SIM), the  
29 sensors *per se*, unfortunately, degrade the original integrity of host composites. Envisaging  
30 such deficiency and facilitated by advances in nanotechnology, we develop a new type of  
31 ultrathin, piezoresistive sensors using spraying coating and nanocomposites formulated with  
32 graphene nanoplatelets/polyvinylpyrrolidone. The sensors are deposited on dielectric  
33 membranes made of partially pre-cured B-stage epoxy films, then electrified using carbon  
34 nanotube film (CNT-film)-made wires, and implanted in carbon fibre-reinforced polymer  
35 composites (CFRPs), to form a sensor network. Only ~45 µm thick (including wires), the  
36 implanted sensors exhibit high compatibility and nonintrusive attributes with CFRPs,  
37 enabling composites to perceive broadband signals *in situ*, ranging from static strain (with a  
38 high gauge factor of 34.5) to structure-guided ultrasonic waves up to 450 kHz – the first time  
39 that piezoresistive sensors implanted in CFRPs respond to dynamic strains in such a broad  
40 frequency band. CFRPs with implanted sensor networks are endowed with capacity of *in*  
41 *situ* SIM, yet not compromising their original integrity. With remarkably reduced intrusion  
42 to composites – as proven in tensile and bending tests, the developed sensors outperform  
43 prevailing sensors for SIM of composites such as lead zirconate titanate-based sensors.

44

45 **Keywords:** CFRPs; structural health monitoring; implantable nanocomposite sensor;  
46 piezoresistive sensor; guided ultrasonic waves

47 **1. Introduction**

48 By virtue of superior mechanical attributes, carbon fibre-reinforced polymer composites (CFRPs)  
49 have attained their maturity to serve as primary load-bearing structural components, in lieu of  
50 conventional metallic materials [1]. As CFRPs age in service, the structural integrity is degraded  
51 progressively, entailing monitoring of structural integrity in a continuous and, preferably, near to a  
52 real-time manner. With years of intensive effort, structural integrity monitoring (SIM) has now  
53 become an essential building block in the overall development of high-performance CFRPs targeting  
54 demanding engineering applications. SIM has proven effectiveness in enhancing the durability of  
55 composites, driving down exorbitant maintenance costs, and extending their residual service life.  
56 With such recognition, the past two decades have witnessed a diversity of SIM techniques that are  
57 tailored for CFRPs [2-5].

58  
59 As the most rudimentary yet pivotal element in SIM, sensors (or elaborately configured sensor  
60 networks) are integrated with CFRPs, for capturing ambient variations and structural changes, on  
61 which basis material degradation or structural damage such as fibre breakage, impact-induced  
62 delamination or matrix cracking can be characterized via appropriate physical models and searching  
63 algorithms. Prevailing sensors that can be used to fulfill SIM for composites broadly embrace acoustic  
64 emission (AE) sensors [6], accelerometers [7], strain gauges [8], fibre Bragg grating (FBG) sensors  
65 [9], interdigital transducers (IDTs) [10], polyvinylidene fluoride (PVDF) film sensors [11], lead  
66 zirconate titanate (PZT) wafers [12, 13], among others.

67  
68 Sensors, of whatever kind, are immobilized with CFRPs via either surface mounting [6, 14] or internal  
69 embedding [9, 15-17], each possessing respective merits and demerits. Depending on applications,  
70 selection of a proper means for sensor immobilization in composites is somewhat debatable. Surface  
71 mounting is preferable when the convenience of sensor installation, ease of maintenance, and low  
72 replacement cost are priority considerations [18], though it is increasingly challenged due to the  
73 inferior accuracy in signal acquisition, because sensors are exposed directly to the environment and

74 prone to ambient noise. This concern is particularly accentuated for CFRPs used in rugged working  
75 conditions such as the composite-made load-bearing structures in aircraft.

76  
77 In contrast to the surface attachment, internal embedment of sensors has advantages when isolation  
78 of sensors from the environment is of necessity. As a consequence of the reduced exposure to ambient,  
79 the embedded sensors usually offer a high signal-to-noise ratio and enhanced immunity to  
80 measurement noise, along with stability, durability and repeatability of long-term signal acquisition.  
81 Internal embedding of sensors is also conducive to preventing detachment of sensors from composites  
82 due to reasons such as adhesive degradation or environmental attack (*e.g.*, moisture and corrosion).  
83 In addition, embedded sensors are in general more sensitive to internal damage than surface attached  
84 sensors are [17]. When sensors, as typified by rigid PZT wafers or brittle FBG sensors, are embedded  
85 in CFRPs, it is envisaged that the sensors may compromise the integrity of the host composites,  
86 regardless of the fact that the intended role of the sensors is to implement integrity monitoring of the  
87 CFRPs [19, 20]. To put it into perspective, Konka *et al.* [21] proved that an embedded PZT wafer  
88 could reduce the interlaminar shear strength of CFRPs remarkably by up to 15%, and in the meantime  
89 observed premature material failure initiated from the PZT wafer edge. This can be attributed to the  
90 fact that the significant difference in mechanical properties between PZT materials and CFRPs  
91 discontinues the strains at the sensor interface and engenders high stress concentration therein.

92  
93 In addition to the embedded sensors themselves, the associated wires/cables, electrodes, and the  
94 dielectric films such as polyimide (Kapton) [15, 16] that are used to insulate sensors from carbon  
95 fibres to prevent short-circuit, also intensify the degradation of structural integrity. The dielectric  
96 films, though thin, can alter the microstructure of a fibre-reinforced matrix, influence the interlaminar  
97 stress distribution in sensor vicinity, introduce artificial defect, and consequently lower the load-  
98 carrying capability of CFRPs. Andreades *et al.* [17] demonstrated that both the flexural and  
99 compression strengths of CFRPs decreased by 8% and 12%, respectively, due to embedded  
100 transducers coated with polyimide films. Xiao *et al.* [6] examined the influence of embedded metallic

101 materials on tensile properties of CFRPs, to reveal that delamination at the interface between the  
102 metal and matrix was a key damage mechanism, implying that embedded metal wires could result in  
103 integrity degradation of composites.

104  
105 Other challenging issues arising from sensor embedding, in comparison with surface mounting, also  
106 include the difficulty in precisely placing and aligning sensors in composites during lay-up, the ease  
107 in breaking vulnerable cables (used for linking sensors to external acquisition equipment) during  
108 edge-trimming, and the impossibility to access or replace embedded sensors during maintenance,  
109 along with high manufacturing cost. Furthermore, PZT wafers, rigid and brittle, are inadaptable to a  
110 curved structure, while FBG sensors are in generally not sensitive to damage far from their vicinity  
111 [22]. Moreover, due to their high sensitivity to moisture and chemicals, FBG sensors need to be  
112 protected by polymer sheaths which may introduce localized mechanical weakness in the composites  
113 [16]. Together, all these factors can significantly undermine the original integrity of CFRPs due to  
114 sensor intrusion, cause premature failure, and jeopardize the use of SIM for enhancing structural  
115 reliability.

116  
117 On the other hand, the recent prosperity in nanotechnology, printed electronics and additive  
118 manufacturing has ushered in a promising avenue to substantially downsize sensors, whereby the  
119 intrusion to composites might be minimized, along with appealing merits such as reduced producing  
120 cost, standardized fabrication procedure, as well as accurate sensor alignment and positioning [23].  
121 As a pioneering study, Thestenson *et al.* [24] dispersed an electrical percolating carbon nanotubes  
122 (CNTs) network in glass fibre-reinforced polymer composites (GFRPs), for detecting matrix-  
123 dominated failures, revealing that an embedded CNT network inflicted insignificant effect on the  
124 integrity of GFRPs. In a similar vein, dip-coating [25], electrophoretic deposition (EPD) [26], or in-  
125 situ chemical vapor deposition (CVD) growth [27] were, respectively, used to decorate glass fibres  
126 with carbonaceous nanomaterials for sensing purposes, while not at the cost of sacrificing the

127 composites' integrity. Nevertheless, these approaches are usually constrained to polymer composites  
128 with dielectric glass fibres, in which the short-circuit of sensors by fibres is not a concern.

129

130 To break through such constriction and extend sensor embedding from GFRPs to CFRPs for strain  
131 measurement, macroscopic carbon nanotube sheets [28, 29] (also referred to as “buckypaper”) are  
132 adopted to avoid short-circuit of sensors by carbon fibres. Exemplarily, Boztepe *et al.* [29] used a  
133 porous CNTs-enriched buckypaper to monitor damage progression in CFRPs under a uniaxial fatigue  
134 load. Analogously, most SIM approaches in this category lie in the premise that damage in  
135 composites, if any, can alter, more or less, the electrical impedance measured by the buckypaper.  
136 However, in addition to the complex fabrication process and difficulty in controlling the initial  
137 electrical properties of the buckypaper, these approaches may fail to provide quantitative evaluation  
138 of the degradation (e.g., accurate location and estimate of severity) – that is because the change in  
139 electrical impedance only holistically reflects the alteration in material between the two electrodes of  
140 the buckypaper – that is global information on the overall change in structural integrity. They may  
141 also overlook initial degradation or embryonic damage, because damage of this degree usually incurs  
142 unmeasurable shift in electrical conductivity of the buckypaper.

143

144 In contrast to scrutinizing global changes in electrical properties, exploration of localized information  
145 such as composite structure-guided ultrasonic waves (GUWs) makes quantitative evaluation of  
146 integrity degradation possible, thanks to the localized information that GUWs carry and accumulate  
147 along wave propagation path. GUWs also offer much higher sensitivity to material degradation of  
148 small scale owing to their short wavelengths, compared with those approaches using global  
149 information (e.g., electrical impedance). In conventional GUV-based methods [13, 16], piezoelectric  
150 sensors (e.g., PZT wafers) are commonly employed to acquire GUWs, which, however, unavoidably  
151 threaten the composites' integrity. As far as nanocomposite-based sensors are concerned, although  
152 they may not inflict remarkable intrusion on composites, this sort of sensors usually fails to respond  
153 to high-frequency GUWs – because GUWs for SIM are of high frequency (of several kHz or even

154 MHz) yet ultralow magnitude (of the order of micro-strain, even nano-strain) – beyond the  
155 responsivity of ordinary nanocomposite-based sensors.

156

157 In conclusion, though significance of SIM for CFRPs cannot be overemphasized, it is still a vast  
158 challenge to implement continuous SIM in a quantitative manner, but without detrimental impact on  
159 composites due to sensor intrusion. Motivated by such recognition and by expanding the authors'  
160 continued endeavour in developing nano-engineered sensors for GUW-based structural health  
161 monitoring [18, 30-36], this study takes advantage of recent technological advances in  
162 nanocomposites and additive manufacturing, aspiring to a new breed of ultrathin, implantable, nano-  
163 engineered piezoresistive sensor, capable of perceiving broadband signals and implementing GUW-  
164 based SIM for CFRPs, yet without compromising the composites' original integration, mechanical  
165 properties and electromechanical performance. The sensors are fabricated using spraying coating and  
166 nanocomposites formulated with graphene nanoplatelets (GNPs) and polyvinylpyrrolidone (PVP).  
167 With high compatibility and nonintrusive attributes with CFRPs, the implanted sensors enable the  
168 composites to perceive broadband signals *in situ*, ranging from static load-induced strain to structure-  
169 guided ultrasonic waves. Results are compared with those when PZT wafers are used as sensors.  
170 Tensile and bending tests are performed in accordance with ASME standards, to examine possible  
171 effect of the implanted sensors on mechanical attributes of CFRPs.

172

## 173 **2. Sensor Fabrication and Morphological Characterization**

### 174 **2.1 Sensing Mechanism**

175 GUWs for SIM are often generated in a range from several hundred kilohertz (kHz) to several  
176 megahertz (MHz), within which GUWs feature a magnitude of the order of microstrain or even  
177 nanostrain. Crucial to faithful acquisition of GUWs is the sensitivity of the sensor material to external  
178 disturbance. To endow sensors with sufficient sensitivity to GUWs of high frequencies and ultralow  
179 magnitudes, the quantum tunneling effect is exploited. The tunneling effect, a quantum mechanical

180 phenomenon, is triggered at nanoscale in a morphologically proper conductive network, when the  
 181 insulative barrier among neighbouring conductive nanoparticles are smaller than a critical threshold  
 182 (of the order of several nanometers in general) [37, 38], under which electrons are able to move  
 183 through these barriers and consequently induce a tunneling current [39]. The percolation threshold  
 184 represents a critical fraction of the conductive nanoparticles in the formed network, beyond which a  
 185 slight increase in the content of conductive nanoparticles can lead to a remarkable leap in the electrical  
 186 conductivity of the material [40]. Because the barrier to break (so that a tunneling current can be  
 187 triggered) is of the order of several nanometers only, it is possible for a GUW, even of ultralow  
 188 magnitude, to alter inter-distances among nanoparticles and activate the tunneling effect, provided  
 189 the nanoparticle-formed network is morphologically appropriate and the content of conductive  
 190 nanoparticles is close to the percolation threshold. A triggered tunneling current alters the  
 191 piezoresistivity manifested by the nanocomposite materials.

192  
 193 Driven by such a quantum tunneling effect-based sensing mechanism, a new type of nanocomposite-  
 194 based piezoresistive sensor is developed. Upon even dispersion of nanoparticles in matrix, the  
 195 electrical properties of the synthesized nanocomposites can be calibrated using the percolation theory  
 196 [40], as

$$197 \quad \sigma \propto (f - f_c)^t, \quad \text{when } f > f_c \quad (1)$$

198 where  $\sigma$  signifies the electrical conductivity of nanocomposites,  $f$  the fraction of nanoparticles,  
 199  $f_c$  the percolation threshold, and exponential  $t$  a constant reflecting the dimensionality of the  
 200 nanocomposites [37, 38]. The electrical resistance of the sensor ( $R$ ) embraces three key components:  
 201 the intrinsic resistance of nanoparticles ( $R_{Nanoparticle}$ ), the contact resistance due to direct contact  
 202 among nanoparticles ( $R_{Contact}$ ), and the tunneling resistance among nanoparticles ( $R_{Tunnel}$ ) that is  
 203 induced by the tunneling effect, namely

$$204 \quad R = R_{Nanoparticle} + R_{Contact} + R_{Tunnel} \cdot \quad (2)$$



205 The tunneling effect is particularly prominent when the content of conductive nanoparticles is close  
206 to the percolation threshold of the nanoparticle-formed conductive network, and therefore  $R_{Tunnel}$   
207 dominates the responsivity of a sensor when the sensor perceives a GUW signal – this is different  
208 from the case when the sensor is subject to an external load with a relatively large magnitude (*e.g.*,  
209 structural vibration or tensile-induced strain) under which the change in electrical resistance of the  
210 sensor is mainly attributed to the breakup of conductive network or loss of contacts among  
211 nanoparticles (i.e., change in  $R_{Contact}$ ).

212

## 213 **2.2 Fabrication of Implantable Sensors**

214 A specific kind of conductive polymer nanocomposites (CPCs) is prepared as the nanocomposite ink  
215 for sensor fabrication using spray coating. The fabricated sensors are further electrified, circuited and  
216 networked with highly conductive CNT-film-made wires, to be implanted into CFRPs and form a  
217 sensor network.

218

### 219 ***Synthesis of Nanocomposite Ink***

220 A standard solution mixing process is applied to prepare the sprayable nanocomposite ink. PVP  
221 (Sigma-Aldrich<sup>®</sup> PVP K-30) is chosen as the matrix, and GNPs (TANFENG<sup>®</sup>, thickness: ~1 nm,  
222 diameter: ~50  $\mu\text{m}$ , SSA: ~1200  $\text{m}^2/\text{g}$ , and purity: >99 wt.%) as the modified nanofillers to form  
223 conductive pathways in CPCs. CPCs thus-produced are dispersed in 20 ml ethanol (Honeywell<sup>®</sup>  
224 24194). The solution is magnetically stirred at 500 rpm at an ambient temperature (25  $^{\circ}\text{C}$ ) for 2 h,  
225 followed with sonication-assisted dispersion (Brandson<sup>®</sup> 5800 Ultrasonic Cleaner, 40 kHz) for 1 h,  
226 to facilitate uniform and even dispersion of GNPs in PVP. 10 ml DI-water is added to augment the  
227 condensation point of solvent to circa -42  $^{\circ}\text{C}$  [41].

228

229 To determine the percolation threshold of CPCs, different GNPs-to-PVP mass ratios, varying from  
230 4.0 to 8.0 wt.%, are investigated, respectively. The electrical resistance (ER) of CPCs is calibrated

231 with a digital graphical sampling multimeter (Keithley® DMM7510). The obtained correlation  
232 between GNPs loading and electrical conductivity ( $\sigma$ ) of the prepared CPCs is shown in **Fig. S1**.  
233 Applying the power-law function linear fitting [34] on Eq. (1), the percolation threshold of the CPCs  
234 is determined to be 4.93 wt.%.

235

### 236 *Deposition of Nanocomposite Ink on B-stage Epoxy Film*

237 The sensors, measuring  $20 \times 3 \text{ mm}^2$  each, are fabricated by spraying the aqueous CPCs onto a  
238 substrate – a partially pre-cured B-stage epoxy film, using an airbrush (HD-130). The scanning speed  
239 of nozzle (5 cm/s), stream pressure (0.35 MPa) and distance of target to nozzle (10 cm) are precisely  
240 controlled during the spraying process to warrant a consistent initial resistance of sensor.

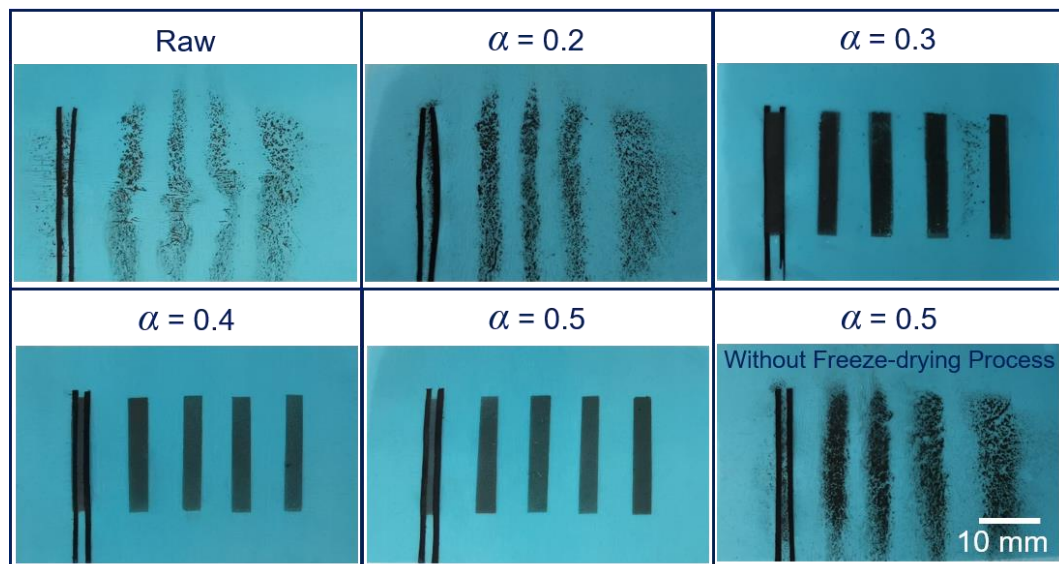
241

242 To make the substrate, B-stage epoxy is spread on a high temperature-resistant release film  
243 (AIRTECH® WL5200B nonperforated peel ply) using a spreader at 80 °C, which is proven effective  
244 to reduce the viscosity of the epoxy and prevent uncontrollable pre-curing. The epoxy film thus-  
245 produced presents the following attributes: 1) the same material properties as the matrix of CFRPs,  
246 to which the sensors are implanted; 2) a strong bonding condition among sensors, carbon fibre  
247 prepregs and CNT-film-made wires (to be detailed in sequent section); and 3) high flexibility  
248 (therefore adaptive to a curved structure) and customizable sensor thickness. These merits warrant  
249 high compatibility and non-invasive attributes of implanted sensors with CFRPs. To limit possible  
250 flowing of the epoxy during curing, the epoxy film is partially pre-cured, whereby to minimize the  
251 effect of epoxy on morphology of the nanocomposite ink deposited on the substrate. The curing  
252 degree of epoxy is determined using the differential scanning calorimetry (DSC) technique (Mettler  
253 Toledo® DSC3). The curing degree is defined as the ratio of the evolved heat  $Q_t$  till time  $t$  to the  
254 total released heat  $Q_{ult}$  during the entire curing process [42], as

255

$$\alpha = \frac{Q_t}{Q_{ult}}, \quad (3)$$

256 The evolved heat flow and corresponding curing degree ( $\alpha$ ) measured by DSC at 130 °C is shown in  
 257 **Fig. S2** and **Table S1** exemplarily lists five groups of measurement of  $\alpha$  against heating duration.  
 258  
 259 During deposition of the nanocomposite ink on the substrate, the epoxy film remains active and  
 260 adhesive, the morphology of which may alter under a thermal load. To evaporate the solvent  
 261 exhaustively, the epoxy film is treated with a freeze-drying process. **Figure 1** compares the  
 262 morphology of a series of nanocomposite inks which are deposited on epoxy films processed with  
 263 different pre-treatments, in which different heating durations are adopted. For comparison, the  
 264 nanocomposite ink is also deposited on an epoxy film which is not processed with such freeze-drying.  
 265 It can be observed that the original morphology of the nanocomposite ink deposited on a raw epoxy  
 266 film without any pre-treatment is not maintained through curing process, as a result of the resin  
 267 flowing. With an increase in pre-curing degree, deposited nanocomposite ink demonstrates improved  
 268 deposition quality, and a good morphology sustains when  $\alpha = 0.4$  afterwards. Also noted in **Fig. 1** is  
 269 that exhaustive evaporation facilitates a precise shaping of the nanocomposite ink on the substrate.



270  
 271 **Fig. 1.** Comparison of ink deposited-epoxy films processed with different pre-treatments.  
 272

273 Another pre-cured B-stage epoxy film is thus produced, and then placed atop the above epoxy film  
 274 on which the nanocomposite sensors have already been deposited. Both films, atop and beneath the  
 275 sensors, respectively called “*upper membrane*” and “*lower membrane*” in what follows, serve as a

276 pair of dielectric membranes, to insulate the sensors from the conductive carbon fibres upon sensor  
277 implantation in CFRPs.

278

### 279 *Fabrication of CNT-film-made Wires*

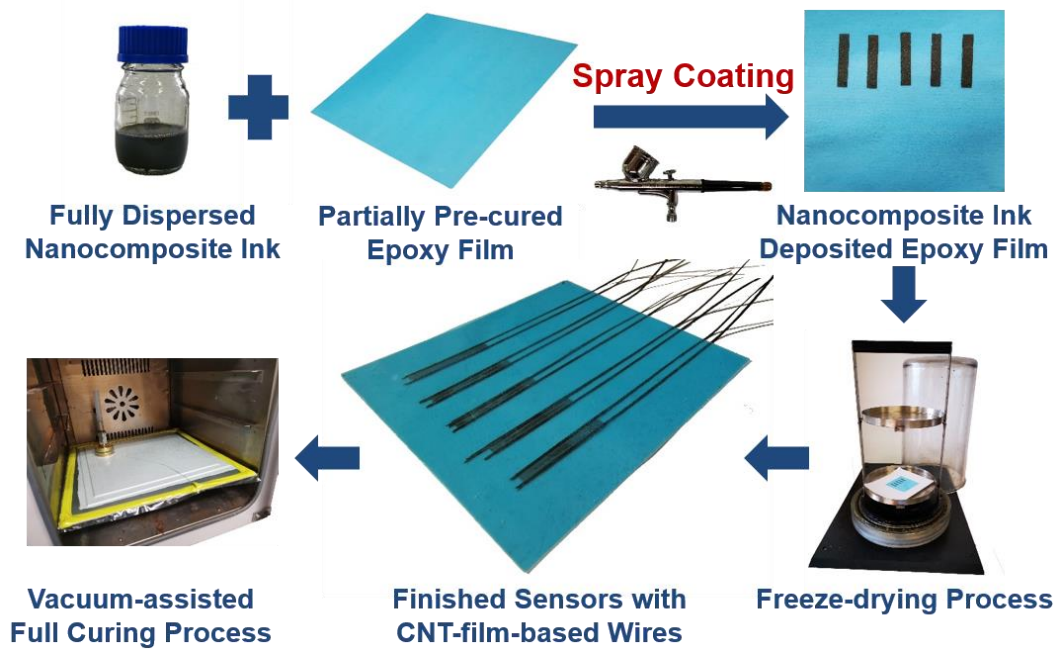
280 To further minimize the intrusion to the host composites when sensors (along with associated wires)  
281 are implanted, a highly conductive wires is fabricated using CNT-film (DexMat<sup>®</sup>) made of pure  
282 CNTs. The wire is tailored ~1 mm in width and ~10 microns in thickness. With key parameters listed  
283 in **Table S2**, the linear resistance of the CNT-film-made wire is only ~20 Ω per meter in length and  
284 1 mm in width, guaranteeing good transmission quality but low intrusion to composites.

285

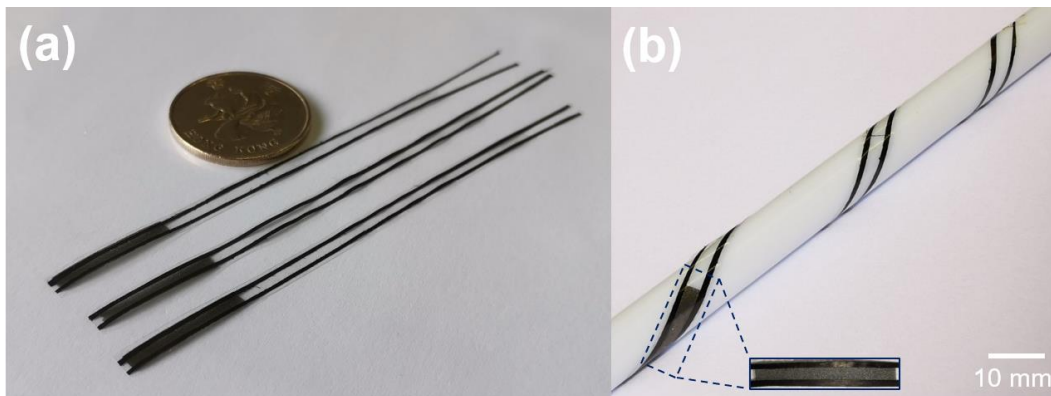
286 Two CNT-film-made wires are aligned along the two opposite edges of each sensor, prior to the  
287 encapsulation by the upper membrane. Thus-encapsulated sensors are then fully cured under a  
288 vacuum-assisted heating condition (130 °C, 60 min). Notably, different from conventional sensors  
289 such as PZT wafers internally embedded in or surface mounted on composites, which are electrified  
290 via a pair of electrodes, the vacuum-assisted curing process in this approach imposes a high pressure  
291 to indent the CNT-film-made wires into nanocomposite sensors, guaranteeing a good conductive  
292 connection but without additional electrodes. Abandoning the conventional electrodes further  
293 minimizes the intrusion of the implanted sensors to the host composites.

294

295 Lightweight, flexible, small and ultrathin, the fabricated sensors can be deployed in a large quantity  
296 in a CFRP structure to configure a dense sensor network. The complete fabrication process of the  
297 sensors is illustrated in **Fig. 2**, and sensors produced in their final fashion are pictured in **Fig. 3**.



**Fig. 2.** Flowchart of fabrication process of implantable sensors.



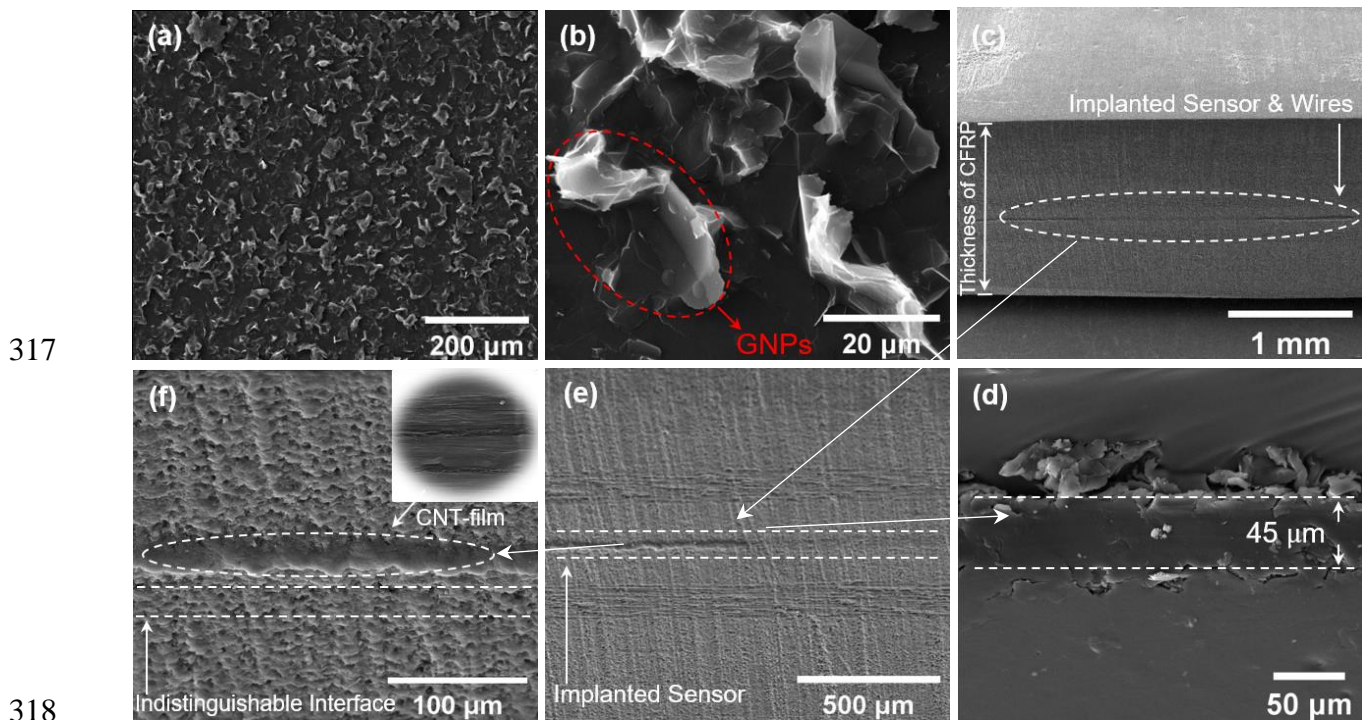
**Fig. 3.** Photographs of (a) finished nanocomposite sensors with CNT-film-made wires; and (b) sensors showing good flexibility (diameter of the plastic rod in photo: 8 mm).

### 2.3 Implantation of Networked Sensors into CFRPs

305 A series of 8-layer quasi-isotropic CFRP laminates of different dimensions, following a stacking  
 306 sequence of  $[0^\circ/90^\circ/45^\circ/-45^\circ]_s$ , are made using unidirectional (UD) CF prepregs (Torayca<sup>®</sup> T300), to  
 307 each of which the fabricated nanocomposite sensors are implanted to form a sensor network, as  
 308 schematically illustrated in **Fig. S3**. An industrial-grade autoclaving (Econoclave<sup>®</sup> EC1.2MX2.4M)  
 309 curing cycle is followed with key parameters listed in **Table S3**. Upon full curing, the nominal  
 310 thickness of each CFRP laminate measures 1.15 mm approximately. Each laminate is then trimmed  
 311 using a water jet cutter (OMAX<sup>®</sup> PROTOMAX).

312 **2.4 Morphological Characterization**

313 To gain an insight into the nanoscale morphology of the fabricated nanocomposite sensors and the  
314 microscopic structure of the sensor-implanted CFRP laminate, morphological characterization is  
315 performed on a scanning electron microscopy (SEM) platform (TESCAN® Vega 3), with some  
316 representative SEM images displayed in **Fig. 4**.



**Fig. 4.** SEM images of (a, b) GNPs/PVP nanocomposite sensor, at two different scales; (c) cross-sectional view of the CFRP laminate with an implanted sensor and wires; (d) individual implantable sensor; (e) vicinity of the implanted sensor; and (f) the interface between the implanted sensor and host composite.

324 It is observed in **Fig. 4(a)** that GNPs are dispersed evenly in PVP matrix, and in **Fig. 4(b)** at a smaller  
325 scale. The even and uniform distribution of GNPs in PVP is a prerequisite to trigger the tunnel current  
326 when a GUV traverses the sensor. **Figure 4(c)** shows the entire cross-section of the CFRP laminate  
327 with an implanted sensor with a CNT-film-made wire. The thickness of the implanted sensor  
328 including its wire measured in **Fig. 4(d)** is  $\sim 45 \mu\text{m}$ . **Figure. 4(e)** shows the vicinity of the implanted  
329 sensor, and at a smaller scale in **Fig. 4(f)**, in both of which it can be noted that the interface between  
330 the composite matrix and the sensor (including wire) is obscure, implying a high degree of  
331 compatibility of the nanocomposite sensor with CFRPs. The inset of **Fig. 4(f)** shows the SEM image



332 of CNT-film. In conclusion, an ignorable degree of intrusion from implanted sensors and their  
333 associate wires to the laminate is noted.

334

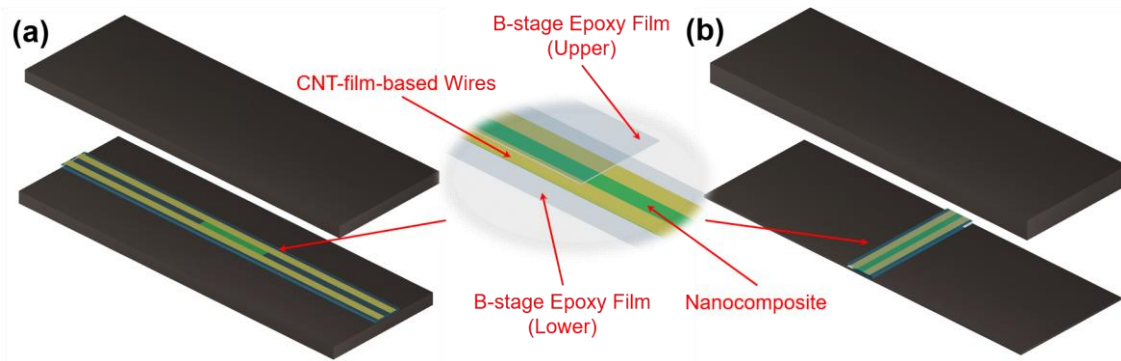
### 335 **3. Mechanical Properties of CFRP Laminates with Implanted Sensors**

336 To scrutinize possible degradation in mechanical properties of the CFRP laminates due to sensor  
337 implantation, two types of mechanical tests – the tensile and flexural tests, are conducted in  
338 accordance with ASTM standards. In each type of test, five CFRP laminates, each implanted with a  
339 fabricated sensor, are prepared in accordance with the manufacturing procedures as described in  
340 Section 2.3, along with another five counterpart laminates of the same dimensions but without any  
341 implanted sensor, for comparison.

342

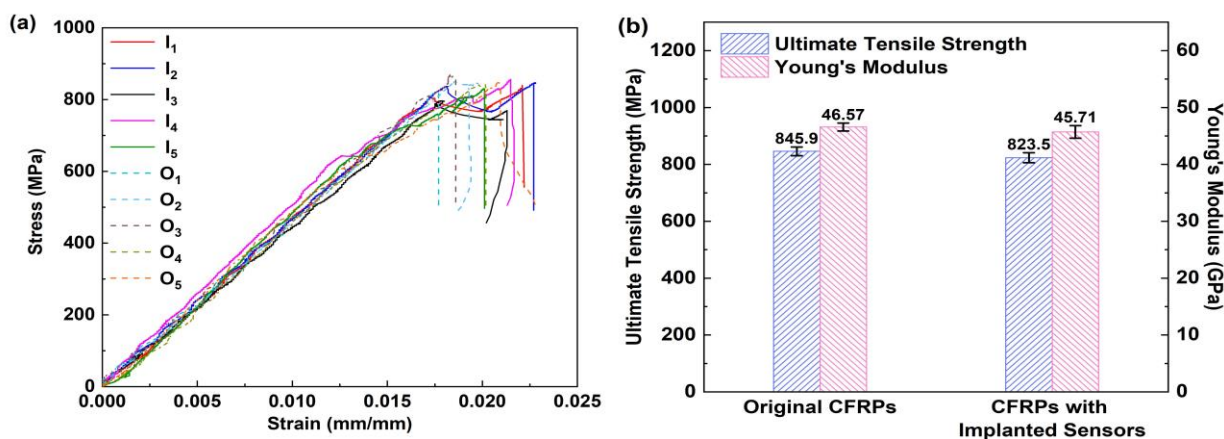
#### 343 **3.1 Tensile Properties**

344 Quasi-static tensile test is performed to calibrate tensile properties of the laminates ( $250 \times 25 \times 1.15$   
345  $\text{mm}^3$ ) and determine their failure modes, according to ASTM D3039. The two ends of each CFRP  
346 laminate are immobilized with an aluminium tab using an adhesive (Scotch-Weld<sup>®</sup> 2216) before tests,  
347 to avoid premature failure near the gripping devices. In each laminate implanted with a sensor, the  
348 sensor ( $250 \text{ mm} \times 3 \text{ mm} \times 45 \mu\text{m}$ ) is centralized in the laminate and electrified via two CNT-film-  
349 made wires (1 mm wide, and  $10 \mu\text{m}$  thick each), as schematically illustrated in **Fig. 5(a)**. The sensor  
350 is positioned between the 4<sup>th</sup> and 5<sup>th</sup> plies of each laminate. All laminates are pulled apart on a  
351 universal testing system (INSTRON<sup>®</sup> 5982) with a crosshead speed of 2 mm/min until fracture, and  
352 strains are simultaneously recorded by an advanced video extensometer (AVE).



353  
 354 **Fig. 5.** Schematic illustration of a CFRP laminate implanted with a nanocomposite sensor (with two  
 355 CNT-film-made wires) for mechanical tests: (a) tensile test; and (b) bending test.

356  
 357 **Figure 6** displays the tensile test results of laminates with and without implanted sensors. In **Fig.**  
 358 **6(a)**, the obtained stress-strain curves argue that the sensor implantation does not incur observable  
 359 degradation in elastic attributes of the CFRP laminate, before reaching its ultimate tensile strength  
 360 (UTS). **Figure 6(b)** compares the averaged UTS and Young's modulus of all laminates with and  
 361 without sensor implantation, confirming no significant change in between. Only the slight reduction  
 362 of 2.65% and 1.85% in the tensile strength and in the tensile stiffness, respectively, is observed,  
 363 attributable to sensor intrusion. In addition, the ignorable degree of standard deviation in test results  
 364 also proves good consistency in sample manufacturing.

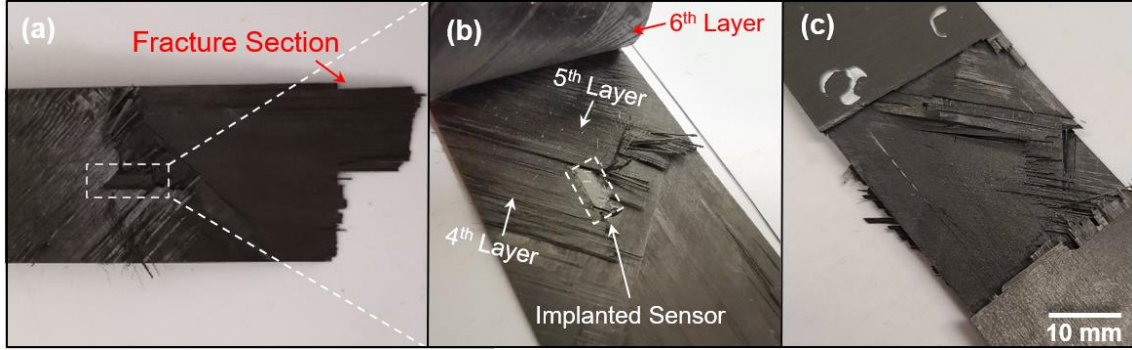


365  
 366 **Fig. 6.** Results of tensile test: (a) stress-strain curves of CFRPs with and without implanted sensors  
 367 (I<sub>1</sub>-I<sub>5</sub>: with implanted sensors; O<sub>1</sub>-O<sub>5</sub>: without any implanted sensor); and (b) averaged UTS and  
 368 Young's modulus of laminates with and without sensor implantation.

369  
 370 The failure type of CFRP laminates with and without implanted sensors is identical, in a mixed mode  
 371 of ply delamination and fibre breakage, as seen in **Fig. 7**. It can be noted in **Fig. 7(a)** and **(b)** that for



372 a laminate with an implanted sensor, the main pathway, along which the delamination is initiated and  
 373 then progresses, is between the 5<sup>th</sup> (-45°) and 6<sup>th</sup> (+45°) plies – analogous to that for those counterpart  
 374 laminates without sensor implantation as observed in **Fig. 7(c)**. In the figures, the fractured regions  
 375 manifest strong bonding between the carbon fibre prepreg and the implanted sensor, clarifying that  
 376 the fracture of the laminate is due to the tensile stress, rather than material degradation at the interface  
 377 between the implanted sensor and the prepreg.



378  
 379 **Fig. 7. (a) A fractured laminate with an implanted sensor; (b) the fracture region in (a); and (c) the**  
 380 **fracture region of a counterpart laminate without sensor implantation.**

381

### 382 3.2 Flexural Properties

383 Three-point bending test is conducted according to ASTM D790, to comparatively examine the  
 384 flexural properties of CFRP laminates with and without implanted sensors. Each laminate measures  
 385  $50.8 \times 12.7 \times 1.15 \text{ mm}^3$ , with a support span of 25.4 mm. A sensor ( $12.7 \text{ mm} \times 3 \text{ mm} \times 45 \text{ }\mu\text{m}$ ) with  
 386 two CNT-film-made wires is implanted at the center of each laminate between the 1<sup>st</sup> and 2<sup>nd</sup> plies,  
 387 at which the maximum normal stress is expected under bending, in **Fig. 5(b)**. The test is also  
 388 conducted on the universal test platform (INSTRON<sup>®</sup> 5982) with a crosshead rate of 0.9 mm/min.  
 389 The maximum normal stress at failure can be determined by **Eq. 4**, while the flexural modulus is  
 390 calculated using **Eq. 5**, by measuring the slope in the linear region of the load-deflection curve, as

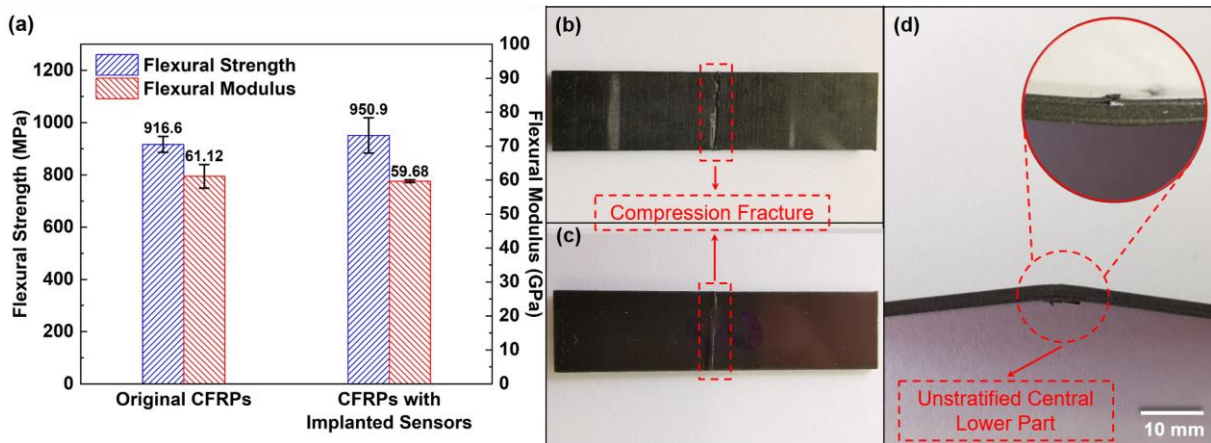
$$\sigma_f = \frac{3PL}{2bd^2}, \quad (4)$$

$$E_B = \frac{L^3m}{4bd^3}, \quad (5)$$

391 where  $\sigma_f$  signifies the flexural strength (MPa),  $P$  the maximum load (N),  $L$  the support span (mm),  
 392  $b$  the width of the laminate (mm),  $d$  its depth (mm),  $E_B$  the modulus of elasticity in bending (MPa)  
 393 and  $m$  the slope of the tangent to the initial straight-line portion of the load-deflection curve.

394

395 The obtained flexural properties of the CFRP laminates with and without implanted sensors are  
 396 compared in **Fig. 8(a)**, to observe nonintrusive effect of an implanted sensor on the flexural properties  
 397 of a CFRP laminate. The flexural strength of the laminate with an implanted sensor deviates from  
 398 916.6 to 950.9 MPa, while flexural modulus decreases marginally from 61.12 to 59.68 GPa. The  
 399 slight variations in both flexural strength (+3.74%) and modulus (-2.36%) are attributed to the  
 400 discrepancy in specimen preparation and tests. **Figures 8(b)-(d)** show the fractured laminates under  
 401 the flexural load, indicating that the failure mode is the compression fracture adjacent to the loading  
 402 nose, irrespective of implantation of a sensor. The implanted sensor does not initiate a defect, from  
 403 which the failure develops, outperforming traditional PZT wafers which inevitably reduce the  
 404 structural integrity to a certain extent [17, 21].



405

406 **Fig. 8.** Results of flexural test: (a) averaged flexural strength and modulus of laminates with and  
 407 without sensor implantation; fracture region of (b) a counterpart laminate without any implanted  
 408 sensor; and (c) a laminate with an implanted sensor, with zoomed-in fracture region in (d).

409

#### 410 4. Dynamic Response Analysis

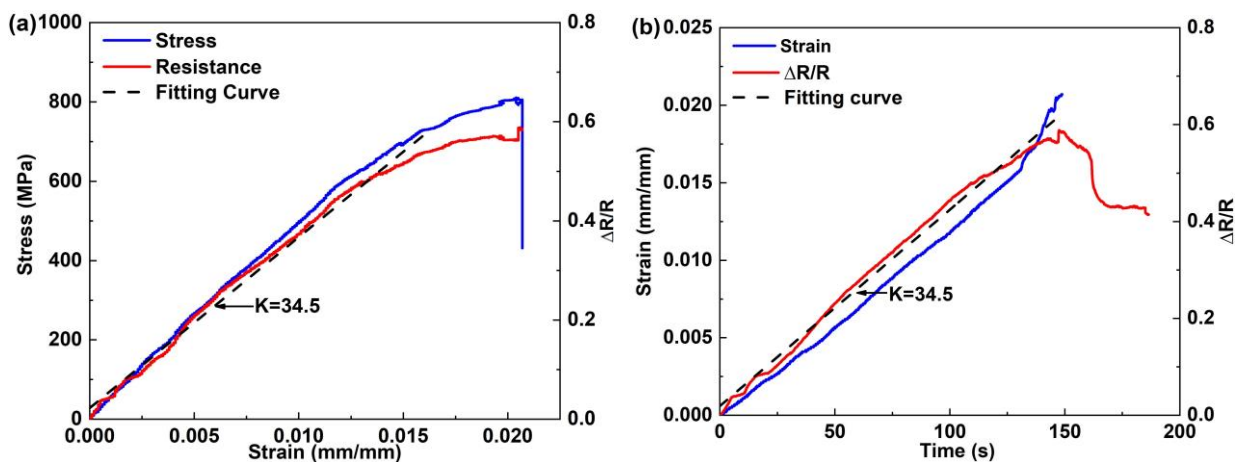
411 With the implanted sensors, dynamic responses of CFRP laminates, when subjected to static load,  
 412 medium-frequency vibration, and GUWs, respectively, are captured *in situ*.

413 **4.1 Quasi-static Strain**

414 An 8-ply quasi-isotropic CFRP laminate ( $250 \times 25 \times 1.15 \text{ mm}^3$ ) with an implanted sensor placed  
 415 between the 1<sup>st</sup> and 2<sup>nd</sup> plies, is prepared and applied with a quasi-static tensile load, as shown in **Fig.**  
 416 **S4**. A digital graphical sampling multimeter (Keithley<sup>®</sup> DMM7510) measures the change in ER of  
 417 the implanted sensor via explored CNT-film-made wire ends during the loading process, as depicted  
 418 in **Figs. S4(b) and (c)**. For comparison and calibration, the normal strain at the laminate cross-section,  
 419 where the sensor is implanted, is measured using the AVE. **Figure 9(a)** presents the obtained stress-  
 420 strain correlation of the laminate, to observe a linear change in ER prior to yielding. To calibrate the  
 421 sensitivity of the sensor, a gauge factor (*GF*) is introduced, which reads

422 
$$GF = \frac{\Delta R}{R_0} / \varepsilon, \quad (6)$$

423 where  $\Delta R = R - R_0$  ( $R$  is the current resistance and  $R_0$  the initial resistance) and  $\varepsilon$  is the strain.  
 424 In **Fig. 9**, *GF* is determined to be 34.5 within the elastic regime, which is much higher than that of a  
 425 commercial strain gauge ( $\sim 2.0$  in general). As seen in **Fig. 9(b)**,  $\Delta R/R$  curve is in consistence with  
 426 the strain curve within the elastic domain, while a large discrepancy is only observed after yielding,  
 427 which can be attributed to the inconsistency of the effective area for AVE measurement, as seen in  
 428 **Fig. S4** – AVE measures the strain between the two-gauge points (25 mm) while the implanted sensor  
 429 reflects the local strain (1 mm gap between two CNT-film-made wires).



430  
 431 **Fig. 9.** (a) Stress/ER change vs. strain; and (b) strain/ER change vs. time under static load.  
 432

433 The piezoresistive behaviour of the nanocomposite sensor manifested during static loading can be  
434 interpreted using the sensing mechanism detailed in Section 2.1. When the strain is relatively large in  
435 magnitude (in the current case), the variation in contact resistance ( $R_{contact}$ ), due to structural change  
436 in the GNP-formed conductive network within the sensor, is dominant over the variation in tunneling  
437 resistance ( $R_{tunneling}$ ) and in intrinsic piezoresistivity of GNPs ( $R_{Nanoparticle}$ ).

438

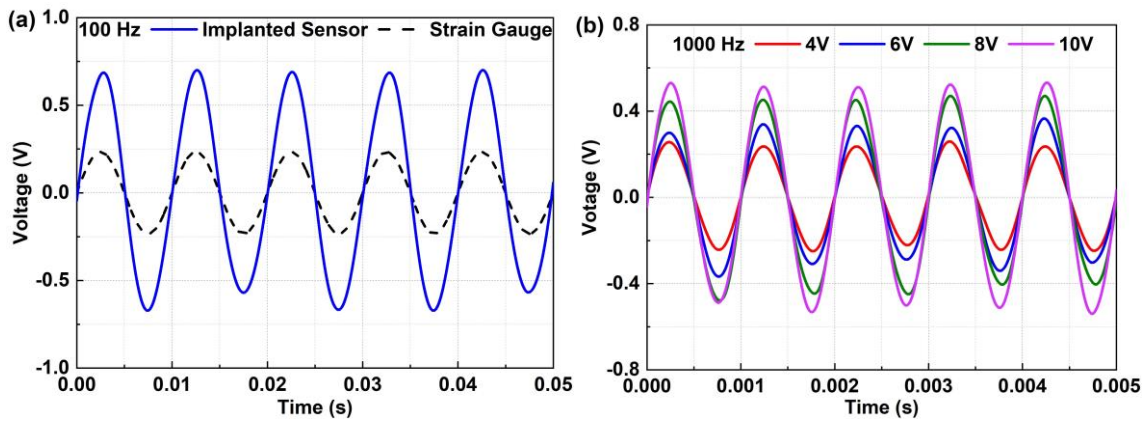
#### 439 **4.2 Medium-frequency Vibration**

440 The responsivity of a CFRP laminate with an implanted sensor to medium-frequency structural  
441 vibration is investigated. An 8-ply quasi-isotropic CFRP laminate ( $250 \times 25 \times 1.15 \text{ mm}^3$ ) with an  
442 implanted sensor is prepared, and one end of the laminate is clamped on a dynamic vibration test  
443 platform as a cantilever beam, as depicted in **Fig. S5**. The sensor is implanted between the 1<sup>st</sup> and 2<sup>nd</sup>  
444 plies of the laminate, 150 mm from the clamped end. A commercial strain gauge with the gauge  
445 resistance of  $120 \Omega$  is surface-glued on the subface (1<sup>st</sup> ply) of the laminate for calibration. Continuous  
446 sinusoidal vibration signals in a sweeping frequency (from 100 to 1000 Hz) are generated with a  
447 waveform generator (HIOKI<sup>®</sup> 7075), to drive a vibration exciter (B&K<sup>®</sup> 4809) which is in contact  
448 with the bottom of the CFRP laminate 35 mm from its free end. Both implanted sensor and the  
449 surface-glued strain gauge are linked to a signal acquisition system which consists of a Wheatstone  
450 bridge (to convert measured piezoresistive variations to electrical voltage signals), a signal  
451 conditioner (KYOWA<sup>®</sup> CDV-900A) and an oscilloscope (Agilent<sup>®</sup> DSO9064A).

452

453 Bandpass filters are applied to the captured raw signals for screening measurement noise. The  
454 processed signals obtained at two typical frequencies, 100 and 1000 Hz, are compared in **Fig. 10**. In  
455 **Fig. 10(a)**, waveform distortion and hysteresis are not observed between signals acquired by the  
456 implanted sensor and by the surface-glued strain gauge. To take a step further, signals captured by  
457 the nanocomposite sensor at 1000 Hz, under different degrees of excitation (4, 6, 8 and 10 V), are  
458 compared in **Fig. 10(b)**, to affirm a linear relationship between the magnitude of excitation and the

459 response intensity of the sensors. Here, it is noteworthy that the nanocomposite sensor developed in  
460 this study is a sort of piezoresistive sensor, and therefore electrical voltage signals, rather than strain  
461 signals, are presented in **Fig. 10**, showing variation in the piezoresistivity measured by the sensor.



462  
463 **Fig. 10.** Responses of CFRPs with an implanted sensor to vibration at (a) 100; and (b) 1000 Hz.  
464

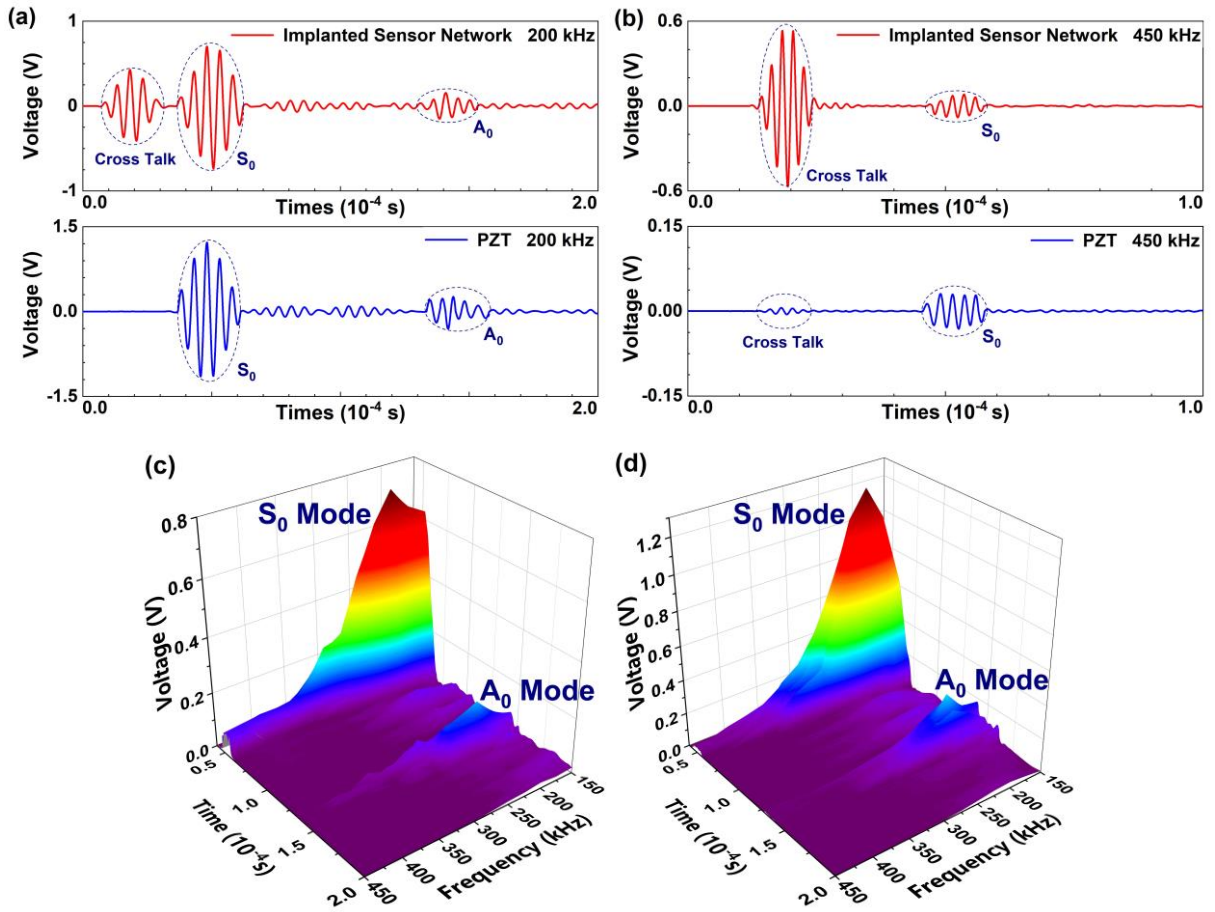
### 465 4.3 High-frequency GUWs

466 To interrogate the responsivity and accuracy of the CFRP laminate with implanted sensors to high-  
467 frequency GUWs, an ultrasonic measurement system is configured, shown schematically in **Fig. S6**.  
468 An 8-ply quasi-isotropic CFRP laminate ( $500 \times 500 \times 1.15 \text{ mm}^3$ ) which is pre-implanted with a  
469 sensor network consisting of four nanocomposite sensors is prepared. The sensor network is  
470 implanted between the 4<sup>th</sup> and 5<sup>th</sup> plies, with each sensor being 100 mm from an edge, respectively.  
471 A miniaturized PZT wafer (PSN-33,  $\varnothing$ : 12 mm, 1 mm thick) is also embedded between the 4<sup>th</sup> and  
472 5<sup>th</sup> plies, alongside of an implanted sensor, which simultaneously perceives GUW signals for  
473 comparison. To emit a GUW into the laminate, a PZT wafer (PSN-33,  $\varnothing$ : 12 mm, 1 mm thick) is  
474 surface-mounted at the center of the laminate. A 5-cycle Hanning-function-modulated sinusoidal  
475 toneburst is generated with a waveform generator (configured on a NI<sup>®</sup> PXIe-10071 platform) and  
476 applied on the surface-mounted PZT wafer, amplified to 400 V<sub>p-p</sub> via a linear power amplifier  
477 (Ciprian<sup>®</sup> US-TXP-3). The GUW propagation signals are perceived by all implanted sensors and the  
478 miniaturized PZT wafer.  
479

480 All captured signals are pre-averaged 1024 times to minimize measurement uncertainty, and  
481 subsequently filtered with a first-order Butterworth filter to mitigate ambient noise. **Figures 11(a)**  
482 **and (b)** compare the signals captured by an implanted nanocomposite sensor that is randomly selected  
483 from the sensor network and by the PZT wafer, when a G UW is respectively activated at 200 and 450  
484 kHz, as two examples, to observe good coincidence in the arrival time for all the wave modes,  
485 including the zeroth-order symmetric Lamb wave mode (denoted by  $S_0$  in the figure), and the zeroth-  
486 order anti-symmetric Lamb wave mode ( $A_0$ ). The discrepancy in signal magnitude is attributed to the  
487 different sensing mechanisms of the two types of sensors: the sensing mechanism of the  
488 nanocomposite sensor is based on piezoresistivity (tunneling effect) as interpreted in **Section 2.1**,  
489 while the PZT wafer measures changes in piezoelectricity. The crosstalk observed in signals at the  
490 excitation moment is generated by the linear power amplifier, which does not interfere with signal  
491 processing and interpretation.

492  
493 Across a broad central frequency range from 150 to 450 kHz (with a step of 25 kHz), all the captured  
494 signals are pre-averaged 1024 times to minimize measurement uncertainty, filtered with a first-order  
495 Butterworth filter to mitigate ambient noise, and consequently processed with a Hilbert transform.  
496 Upon signal processing, signal spectra are obtained and shown in **Figs. 11(c)** and **(d)** in a band from  
497 150 to 450 kHz. It shows comparable sensing performance between the implanted sensors and the  
498 PZT wafer. Such a sensing capability enables G UW-based SIM for the host composite structure – a  
499 difficult task to be fulfilled using conventional nanocomposite piezoresistive sensors [28-34]. In  
500 addition, the excitation – a 5-cycle sinusoidal toneburst, is modulated with a Hanning window, so as  
501 to constrain the excited wave energy in a narrow band around the excitation frequency (i.e., 200 kHz  
502 in this study).





503

504

505 **Fig. 11.** Signals captured by a nanocomposite sensor randomly selected from the implanted sensor  
 506 network and by the PZT wafer, at (a) 200 kHz; and (b) 450 kHz; Spectra over time-frequency  
 507 domain up to 450 kHz, perceived by (c) nanocomposite sensor; and (d) PZT wafer.

508

509 **5. Conclusion**

510 A new breed of implantable, nanocomposite-based piezoresistive sensor is developed using an  
 511 additive manufacturing approach (spray coating) with GNPs/PVP-based nanocomposite ink. The  
 512 sensors are encapsulated by dielectric membranes made of partially pre-cured B-stage epoxy films,  
 513 and electrified via CNT-film-made wires. Remarkably different from conventional embeddable  
 514 sensors such as PZT wafers or FBG sensors that unavoidably degrade integrity of the host composites,  
 515 the developed sensors (including wires), only  $\sim 45 \mu\text{m}$  in thickness, exhibit high compatibility and  
 516 nonintrusive attributes with the host composites. Comprehensive tensile and bending tests in  
 517 accordance with ASTM standards demonstrate that the sensors inflict ignorable degradation in  
 518 mechanical properties of the host composites. With an optimal morphology and based on a quantum

519 tunneling effect-based sensing mechanism, the sensors have a sensing capability to perceive  
520 broadband signals *in situ*, ranging from static load-induced strain to structure-guided ultrasonic waves  
521 up to 450 kHz – the first time that piezoresistive sensors implanted in CFRPs responds to dynamic  
522 strains in such a broad frequency band, outperforming traditional nanocomposite-based piezoresistive  
523 sensors which response to low-frequency and large-scale strain only. A gauge factor of 34.5 is  
524 achieved for the sensors when used to acquire static load-induced strain. Featuring merits such as  
525 lightweight, flexibility and compatibility, the sensors can be densely networked and implanted into  
526 CFRPs to implement GUW-based SIM, yet not compromising the composites' original integrity.

527

## 528 **Acknowledgments**

529 The work was supported by a General Project (No. 51875492) and a Key Project (No. 51635008)  
530 received from the National Natural Science Foundation of China. Z Su acknowledges the support  
531 from the Hong Kong Research Grants Council via General Research Funds (Nos. 15202820,  
532 15204419 and 15212417).

533

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