1	Ameliorated-Multiple Signal Classification
2	(Am-MUSIC) for Damage Imaging
3	Using A Sparse Sensor Network
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25 Abstract

26 Multiple Signal Classification (MUSIC) – a directional scanning and searching algorithm, has 27 gained its prominence in phased array-facilitated nondestructive evaluation. Nevertheless, 28 prevailing MUSIC algorithms are largely bound up with the use of a dense linear array, which fail 29 to access the full planar area of an inspected sample, leaving blind zones to which an array fails to scan, along with the incapability of differentiating multiple damage sites that are close one from 30 31 another. To break above limitations, conventional MUSIC algorithm is ameliorated in this study, 32 by manipulating the signal representation matrix at each pixel using the excitation signal series, 33 instead of the scattered signal series, which enables the use of a sparse sensor network with 34 arbitrarily positioned transducers. In the ameliorated MUSIC (Am-MUSIC), the orthogonal 35 attributes between the signal subspace and noise subspace inherent in the signal representation 36 matrix is quantified, in terms of which the Am-MUSIC yields a full spatial spectrum of the 37 inspected sample, and damage, if any, can be visualized in the spectrum. Am-MUSIC is validated, 38 in both simulation and experiment, by evaluating single and multiple sites of damage in plate-like waveguides with a sparse sensor network. Results verify that i) detectability of Am-MUSIC-39 40 driven damage imaging is not limited by damage quantity; ii) Am-MUSIC has full access to a 41 sample, eliminating blind zones; and iii) the amelioration expands conventional MUSIC from 42 phased array-facilitated nondestructive evaluation to health monitoring using built-in sparse 43 sensor networks.

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Keywords: structural health monitoring; guided ultrasonic waves; multiple signal classification
(MUSIC); phased array; sparse sensor network

48 **1. Introduction**

49 With the incentive to "visualise" hidden material defect or structural damage, continued effort has 50 been made to projecting identified results, by means of proper imaging algorithms, in synthetic 51 illustration, in which anomaly, if any, can be imaged intuitively [1]. Amongst various anomaly 52 imaging approaches, those by virtue of guided ultrasonic waves (GUWs) have demonstrated eminent detectability, accuracy and precision [2-5], as represented by tomography-based imaging 53 54 [6], delay-and-sum imaging [7], time-reversal focusing imaging [8], probability-based imaging [9], and array signal processing-based imaging [10-14], to name a few. Amongst them, the array 55 56 signal processing-based imaging can be implemented in various modalities, including sparse 57 reconstruction [10], minimum variance distortionless response method [11], subspace fitting [12], 58 maximum-likelihood method [13], and Multiple Signal Classification (MUSIC) [14].

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60 In particular, the MUSIC algorithm, with its theoretical framework shaped by Schmit [15] in 1981 61 for frequency estimation and radio direction finding, is a directional scanning and searching 62 method to unbiasedly estimate signal features in terms of the orthogonal attributes between signal 63 subspace and noise subspace. With a directional scanning ability, MUSIC has been extended to 64 various application domains such as radar positioning [16], sonar [17], seismic exploration [18], 65 biomedicine [19] and so forth. MUSIC has also proven effectiveness in GUW-based damage imaging. Representatively, Stepinski and Engholm [20] revamped a conventional MUSIC 66 67 algorithm for estimating the direction of arrival (DOA) of incoming waves in plates with a uniform circular array. Yang et al [21, 22] employed the MUSIC algorithm to calculate the arrival 68 69 times of impact-induced waves in conjunction with the use of wavelet transform, whereby to 70 predict the location of an impact to a plate. Majority of MUSIC-driven damage evaluation lies in 71 the far-field hypothesis that assumes a wave scatterer (e.g., damage) within the inspection region 72 is sufficiently far from the phased array, so that the waves emanating from the scatterer can be 73 considered as a plane wave when they arrive at the array, as illustrated schematically in **Fig. 1(a)**. 74

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Fig. 1. Use of phased array for evaluation of damage in a planar inspection region [23]: (a) far-field scenario; and (b) near-field scenario (d_k : the distance from the wave source through the damage then to the k_{th} array element).

89 Despite demonstrated effectiveness in numerous applications, the *far-field hypothesis* is often 90 questioned in practical implementation, as the damage location is unknown a priori, and the wavefront scattered by the damage is naturally cylindrical rather than planar, provided the damage 91 92 is in the near-field – the scenario in **Fig. 1(b)**. In recognition of such deficiency that prevailing 93 MUSIC algorithms inherently has, enhancement has been made to improve conventional MUSIC-94 driven damage evaluation. Zhong et al [23, 24] developed a revamped MUSIC algorithm based 95 on the Taylor expansion theory, applicable to detecting near-field damage in a composite oil tank. 96 Extending this study, Yuan et al [25] took into account the anisotropy of composite structures and 97 proposed a single-frequency component-based MUSIC algorithm able to improve the precision 98 of locating a near-field impact site in composites. Zuo et al [26] calculated the cross-correlation 99 function for the scattered signals received by a damage scattering model and the residual signals 100 received in experiment, and applied the two-dimensional (2-D) MUSIC algorithm to identify 101 damage in plate-like structures. Bao et al [27] combined the transmitter beamforming and 102 weighted image fusion to enhance the conventional MUSIC, endowing it with the capability of 103 localizing near-field corrosion in aluminum plates. Bao et al [28] further proposed a compensated 104 MUSIC algorithm by considering the effect of both the localization error caused by structural 105 anisotropy and the sensor position error, showing improved detection accuracy.

107 Notwithstanding the forgoing, the prevailing MUSIC-based damage imaging approaches in108 general present the following limitations:

- 109 use of linear phased arrays: the MUSIC algorithm, as an eigen-structure mathematic i) 110 approach, is not inherently restricted to the use of linear arrays. When used for damage 111 identification, a linear array that features a dense configuration of transmitter elements is 112 usually adopted, and the element pitch (1) shall be uniform and small enough, ideally 113 satisfying $l \le \lambda / 2$ (λ : the wavelength of the wave generated by the array [29]), to facilitate construction of the signal representation matrix with the scattered signal series. 114 115 In general, the conventional MUSIC algorithms do not sustain the use of a sparse sensor 116 network with individual sensors at arbitrary locations;
- 117 ii) *existence of blind zone*: the beamforming capacity of the algorithm degrades when the 118 scanning angle is close to 0° or 180° . In most circumstances, those regions, where the 119 scanning angles are in the range of [0, 30] or [150, 180], are deemed blind zones [30], in 120 which damage, if any, may be overridden;
- iii) *ambiguous results due to mirror effect*: the identified damage using a MUSIC algorithm
 might be a mirrored dummy of the true damage which is located symmetrically with
 regard to the array surface [31]; and
- iv) obscure multiple damage sites: when MUSIC is used for imaging multiple damage sites,
 the number of scatterers shall be predicted beforehand. To this end, a threshold is selected,
 and the eigenvalues of the covariance matrix of the received signals that are larger than
 this threshold shall be counted as the number of the scatterers. However, selection of the
 threshold is a highly subjective manner at the discretion of individuals, and it is prone to
 contamination of measurement noise [32]. This results in inferior imaging resolution and
 makes it challenging to differentiate multiple damage sites that are close one from another.
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132 Aimed at surmounting the above limitations that prevailing MUSIC-based damage imaging 133 approaches may encounter, an ameliorated MUSIC (Am-MUSIC) algorithm is developed by 134 manipulating the signal representation matrix at each image pixel using the excitation signal series 135 instead of the scattered signal series. Thanks to that, Am-MUSIC algorithm does not necessarily 136 entail the use of a linear phased array, and instead it is compatible with a sparse sensor network 137 in which individual transducers can be positioned arbitrarily. At each image pixel, the orthogonal 138 attributes between the signal subspace and noise subspace inherent in signal representation matrix is quantified, in terms of which Am-MUSIC yields a full spatial spectrum of the inspected sample, 139 140 to visualize damage, regardless of its quantity in the sample. Both simulation and experiment are

141 performed to validate the Am-MUSIC algorithm, by evaluating single and multiple sites of 142 damage in plate-like waveguides with a sparse sensor network comprising only a handful of 143 miniaturized lead zirconate titanate (PZT) wafers.

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The rest of this paper is organized as follows. The conventional MUSIC-driven damage imaging, based on the near-field hypothesis, is briefed in Section 2, on which basis the Am-MUSIC algorithm is developed, with key amelioration detailed in Section 3. Numerical verification of Am-MUSIC is illustrated in Section 4, followed with experimental validation in Section 5.

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150 2. Near-Field MUSIC Algorithm

GUWs guided by a plate-like waveguide, a.k.a. *Lamb waves*, are of a multimodal and dispersive nature. At a given frequency, Lamb waves feature a multitude of wave modes which can be classified as the symmetric and antisymmetric modes. We consider a pure, monochromatic Lamb wave mode in the waveform of a toneburst, as the excitation signal s(t). s(t) is defined in a complex domain as

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$$s(t) = u(t) \exp^{i\omega_0 t}, \tag{1}$$

157 where u(t) denotes a window function to regulate the toneburst, *t* the time, *i* the imaginary unit, 158 and ω_0 the central frequency of the toneburst. With the attenuation in magnitude as wave 159 propagation in consideration, the Lamb wave, $R(\omega)$, after travelling the distance *d* can be 160 represented, in the frequency domain, as

161
$$R(\omega) = \frac{d_0}{\sqrt{d}} S(\omega) \exp^{-ikd}.$$
 (2)

162 In the above, d_0 signifies an initial distance with regard to which the wave attenuation is calibrated; 163 $S(\omega)$ is the corresponding Fourier representation of s(t); $k = \frac{\omega_0}{c}$, where k denotes the 164 wavenumber and c represents the propagation velocity of the considered monochromatic Lamb 165 wave mode.

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167 Substituting Eqs. (1) into (2), the Lamb wave r(t) when it arrives at the distance d can be 168 yielded, in the time domain, as

169
$$r(t) = \frac{d_0}{\sqrt{d}} F^{-1} \left\{ s(\omega) \exp^{-i\frac{w_0}{c}d} \right\} = \frac{d_0}{\sqrt{d}} s(t - \frac{d}{c}) = \frac{d_0}{\sqrt{d}} u(t - \frac{d}{c}) \exp^{i\omega_0(t - \frac{d}{c})}, \quad (3)$$

170 where r(t) is the inverse Fourier transform of $R(\omega)$ and F^{-1} is the inverse Fourier transform.

For an intact waveguide, the captured wave signal, denoted with $r^{\text{measured-intact}}(t)$, is the direct arrival wave $r^{\text{direct}}(t)$, boundary-reflection wave $r^{\text{boundary-reflection}}(t)$ with incoherent noise $w^{\text{measured-intact}}(t)$, as

(4)

175
$$r^{\text{measured-intact}}(t) = r^{\text{direct}}(t) + r^{\text{boundary-reflection}}(t) + w^{\text{measured-intact}}(t),$$

176 where $r^{\text{direct}}(t)$ is the arrival wave propagating along the path from the wave source to the wave 177 receiver. Provided damage is present at an unknown location in the waveguide, the damage can 178 be modeled as a secondary wave source to scatter the incoming Lamb waves. Ignoring mode 179 conversion that is fairly weak in magnitude, the measured signal $r^{\text{measured-damage}}(t)$ comprises the 180 direct arrival wave $r^{\text{direct}}(t)$, boundary-reflection wave $r^{\text{boundary-reflection}}(t)$, additional scattered wave 181 from the damage $r^{\text{scattered}}(t)$, and the incoherent noise $w^{\text{measured-damage}}(t)$, as

182
$$r^{\text{measured-damage}}(t) = r^{\text{direct}}(t) + r^{\text{boundary-reflection}}(t) + r^{\text{scattered}}(t) + w^{\text{measured-damage}}(t), \quad (5)$$

183 where $r^{\text{scattered}}(t)$ is the arrival wave propagating along a scattered path (namely, the path from the 184 wave source to the damage and then to the wave receiver). Suppose that the direct waves and 185 boundary-reflection waves are the same at $r^{\text{measured-intact}}(t)$ and $r^{\text{measured-damage}}(t)$, $r^{\text{scattered}}(t)$ which 186 carries information pertaining to the damage location can be obtained through benchmarking 187 reference signals obtained from the intact status, as

188 $r^{\text{measured-damage}}(t) - r^{\text{measured-intact}}(t) = r^{\text{scattered}}(t) + w(t) = r^{\text{residual}}(t), \quad (6)$

189 where w(t) is the difference between the two noise terms $w^{\text{measured-intact}}(t)$ and $w^{\text{measured-damage}}(t)$ in 190 the intact and current statuses. Here, for convenience of discussion, the terms of $r^{\text{scattered}}(t) + w(t)$ 191 is referred to as the *residual signal*.

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With the near-field assumption, as schematically illustrated in **Fig. 1(b)**, Lamb wave is excited at a foreknown position *P*, scattered by the damage, and then received by a linear sensor array that consisting of *K* transducing elements with a uniform element spacing *l*. According to Eq. (3), the scattered signal received by the first array element, $r_1^{\text{scattered}}(t)$, is

197
$$r_1^{\text{scattered}}(t) = \frac{d_0}{\sqrt{d_1}} s(t - \frac{d_1}{c}) = \frac{d_0}{\sqrt{d_1}} u(t - \frac{d_1}{c}) \exp^{i\omega_0(t - \frac{d_1}{c})},$$
(7)

where d_1 signifies the distance from the wave source through the damage then to the first array element. Let $\tau_k = \frac{d_1 - d_k}{c}$ (*i.e.*, the time delay between two arrival signals captured by the first and 200 the k^{th} (k=1, 2, ..., K) element in the array), and then the scattered wave signal received by the 201 k^{th} element, $r_k^{\text{scattered}}(t)$, can be expressed as

202
$$r_{k}^{\text{scattered}}(t) = \frac{d_{0}}{\sqrt{d_{k}}} s(t - \frac{d_{k}}{c}) = \frac{d_{0}}{\sqrt{d_{k}}} u(t - \frac{d_{k}}{c}) \exp^{i\omega_{0}(t - \frac{d_{k}}{c})} = \frac{d_{0}}{\sqrt{d_{k}}} u(t - \frac{d_{1}}{c} + \tau_{k}) \exp^{i\omega_{0}(t - \frac{d_{1}}{c} + \tau_{k})}.$$

$$(k = 1, 2, K, K)$$

203

With the assumption that the array element spacing *l* is sufficiently small (namely, $l \le \lambda/2$, where λ is the wavelength of wave signal), $r_k^{\text{scattered}}(t)$ can be obtained based on the first element scattered signal $r_1^{\text{scattered}}(t)$ (defined in Eq. (7)) as

(8)

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$$r_{k}^{\text{scattered}}(t) = \frac{d_{0}}{\sqrt{d_{k}}} u(t - \frac{d_{1}}{c} + \tau_{k}) \exp^{i\omega_{0}(t - \frac{d_{1}}{c} + \tau_{k})}$$

$$\approx \frac{d_{0}}{\sqrt{d_{k}}} u(t - \frac{d_{1}}{c}) \exp^{i\omega_{0}(t - \frac{d_{1}}{c} + \tau_{k})}$$

$$= \sqrt{\frac{d_{1}}{d_{k}}} r_{1}^{\text{scattered}}(t) \exp^{i\omega_{0}\tau_{k}}.$$
(9)

According to the cosine theorem [33] and second-order Taylor expansion [34], τ_k can be rewritten as

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$$\tau_{k} = \frac{d_{1} - d_{k}}{c} = \frac{d_{1} - \sqrt{d_{1}^{2} + (k - 1)^{2} l^{2} - 2d_{1}(k - 1)l\cos(90^{\circ} - \theta)}}{c}$$

$$= \frac{-l\sin\theta}{c}(k - 1) + (\frac{-l^{2}}{cd_{1}}\cos\theta^{2})(k - 1)^{2} + O(\frac{l^{2}}{d_{1}^{2}}),$$
(10)

211 where $O(\frac{l^2}{d_1^2})$ denotes those terms, the order of which is greater than or equal to $\frac{l^2}{d_1^2}$. Using the

second-order Taylor series approximation, the scattered wave signal received by the k^{th} element retreats to

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$$r_{k}^{\text{scattered}}(t) = \sqrt{\frac{d_{1}}{d_{k}}} r_{1}^{\text{scattered}}(t) \exp^{i\omega_{0}\tau_{k}} = \sqrt{\frac{d_{1}}{d_{k}}} r_{1}^{\text{scattered}}(t) \exp^{i\omega_{0}(\frac{-l\sin\theta}{c}(k-1)+(\frac{-l^{2}}{cd_{1}}\cos\theta^{2})(k-1)^{2})}.$$
 (11)

215 Letting
$$b_k(d,\theta) = \sqrt{\frac{d_1}{d_k}} \exp^{i\omega_0(\frac{-l\sin\theta}{c}(k-1) + (\frac{-l^2}{cd_1}\cos\theta^2)(k-1)^2)}$$
, as the array steering factor for the k^{th}

scattered signal, and recalling the noise term in Eq. (6), the k^{th} residual signal, $r_k^{\text{residual}}(t)$, can be expressed as

218
$$r_k^{\text{residual}}(t) = b_k(d,\theta) r_1^{\text{scattered}}(t) + w_k(t).$$
(12)

For the linear array with *K* elements, the residual signal vector $\mathbf{R}^{\text{residual}}(t)$ can thus be obtained and expressed in a signal representation matrix, which reads

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$$\mathbf{R}^{\text{residual}}(t) = \mathbf{B}(d,\theta)r_1^{\text{scattered}}(t) + \mathbf{W}(t), \qquad (13)$$

222 where

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$$\mathbf{R}^{\text{residual}}(t) = [r_1^{\text{residual}}(t), \mathbf{L}, r_k^{\text{residual}}(t), \mathbf{L}, r_K^{\text{residual}}(t)]^T,$$

$$\mathbf{B}(d,\theta) = [b_1(d,\theta), \mathbf{L}, b_k(d,\theta), \mathbf{L}, b_K(d,\theta)]^T$$

$$= \begin{bmatrix} 1 \\ M \\ \sqrt{\frac{d_1}{d_k}} \exp^{i\omega_0(\frac{-l\sin\theta}{c}(k-1) + (\frac{-l^2}{cd_1}\cos\theta^2)(k-1)^2)} \\ M \end{bmatrix},$$
224

 $\mathbf{W}(t) = [w_1(t), \mathbf{L}, w_k(t), \mathbf{L}, w_k(t)]^T$.

 $\sqrt{\frac{d_1}{d_K}} \exp^{i\omega_0(\frac{-l\sin\theta}{c}(K-1)+(\frac{-l^2}{cd_1}\cos\theta^2)(K-1)^2)}$

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Prevailing MUSIC-based damage imaging approaches have been developed by virtue of the signal
representation matrix as defined in Eq. (13). They, in general, present the following limitations
during practical implementation, as preliminarily commented in the preceding section:

i) In Eq. (9), the operation of approximation, u(t - d₁/c + τ_k) ≈ u(t - d₁/c), lies in the premise that
τ_k is negligibly small. To accommodate such a pre-requisite, the element spacing in the
phased array must be sufficiently small (l ≤ λ / 2), leading to a dense configuration of the
transducing elements; and
ii) In Eq. (11), the steering vector is approximated using the second-order Taylor approximation,
and the range error introduced by such approximation is remarkable when the damage is close
to the array. For a range that is smaller than twice the array length (i.e., the length from the

first element to the k^{th} element), such error could be 10% or above due to such approximation [35]. In addition, the steering vectors at the scanning angles θ and $180^{\circ} - \theta$ have the same value in Eq. (11), resulting in ambiguous results due to mirror effect.

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241 **3. Am-MUSIC with A Sparse Sensor Network**

242 Aimed at circumventing the above key limitations that conventional MUSIC-based damage

imaging possesses, the original MUSIC algorithm is revamped. Different from the use of a linear 243 244 phased array, we allow a sparse sensor network with individual transducers that are randomly positioned. Without loss of generality, consider a sparse sensor network comprising Q PZT wafers 245 246 (labelled as PZT-1, PZT-2, ..., PZT-j, ..., PZT-Q), as shown in Fig. 2. Positioned at an arbitrary 247 location within the inspection region, each PZT wafer acts as either a wave transmitter or a wave 248 receiver, leading to M = Q(Q-1)/2 transmitter-receiver pairs in the sensor network. Provided damage exists at pixel (x, y) within the inspection area, the propagation distance, d_{my} , for a 249 Lamb wave, which is generated by the i^{th} transmitter at (x_i, y_i) , scattered by damage at (x, y)250 and then propagates to the j^{th} receiver at (x_j, y_j) , is 251

252
$$d_{mxy} = \sqrt{(x - x_i)^2 + (y - y_i)^2} + \sqrt{(x - x_j)^2 + (y - y_j)^2} = c \cdot t_{mxy}, \qquad (14)$$

253 and t_{mxy} is the time for the wave traveling along the scattered path.



254 255

Fig. 2. A plate waveguide with a sparse sensor network of Q PZT wafers.

Therefore, the scattered signal received by the m^{th} transmitter-receiver pair, $r_m^{\text{scattered}}(t)$, can be written according to Eq. (3) as

258
$$r_m^{\text{scattered}}(t) = \frac{d_0}{\sqrt{d_{mxy}}} s(t - \frac{d_{mxy}}{c}) \qquad (m = 1, 2, L, M).$$
(15)

Equation (15) argues that for *M* transmitter–receiver pairs rendered by the sensor network, different scattering paths feature different degrees of time delay. A time shift, t_{mxy} , is then applied to the m^{th} scattered signal $r_m^{\text{scattered}}(t)$ in Eq. (15), as

262
$$r_m^{\text{scattered}}(t+t_{mxy}) = \frac{d_0}{\sqrt{d_{mxy}}} s(t - \frac{d_{mxy}}{c} + t_{mxy}) = \frac{d_0}{\sqrt{d_{mxy}}} s(t).$$
(16)

263 Letting $a_{mxy} = \frac{d_0}{\sqrt{d_{mxy}}}$ (a_{mxy} is referred to as the array steering factor for the m^{th} scattered signal

in what follows), Eq. (16) can be rewritten as

265
$$r_m^{\text{scattered}}(t+t_{mxy}) = a_{mxy}s(t).$$
(17)

With the noise term (w(t) in Eq. (6)) in consideration, the residual signal vector for a total of Mreceived signals which are respectively scattered by the damage at pixel (x, y), $\mathbf{R}_{xy}^{\text{residual}}(t)$, can be expressed as the signal representation matrix

$$\mathbf{R}_{xy}^{\text{residual}}(t) = \mathbf{A}_{xy}s(t) + \mathbf{W}(t), \qquad (18)$$

where

271
$$\mathbf{R}_{xy}^{\text{residual}}(t) = [r_1^{\text{residual}}(t+t_{1xy}), \mathbf{L}, r_m^{\text{residual}}(t+t_{mxy}), \mathbf{L}, r_M^{\text{residual}}(t+t_{Mxy})]^T$$

272

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273
$$\mathbf{W}(t) = [w_1(t+t_{1y}), \mathbf{L}, w_m(t+t_{my}), \mathbf{L}, w_M(t+t_{My})]^T$$

Equation (18) implies that after compensating for the time delay to each residual signal, the residual signal vector can be defined using the excitation signal series, instead of using the scattered signal series as a conventional MUSIC algorithm does (Eq. (13)). It is such a merit of the ameliorated MUSIC (Am-MUSIC) algorithm that enables the use of a sparse sensor network with arbitrarily positioned transducers.

 $\mathbf{A}_{xy} = [a_{1xy}, \mathbf{L}, a_{mxy}, \mathbf{L}, a_{Mxy}]^T,$

279

Recalling the MUSIC algorithm, the covariance matrix C of the residual signal vector at pixel (x, y) within the inspection region yields as

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$$\mathbf{C} = E[\mathbf{R}_{xy}^{\text{residual}}(t)\mathbf{R}_{xy}^{\text{residual}}(t)^{H}]$$

= $\mathbf{A}_{xy}E[s(t)gs(t)^{H}]\mathbf{A}_{xy}^{H} + \mathbf{A}_{xy}E[s(t)g\mathbf{W}(t)^{H}] + E[\mathbf{W}(t)gs(t)^{H}]\mathbf{A}_{xy}^{H} + E[\mathbf{W}(t)g\mathbf{W}(t)^{H}],$
(19)

where E[] denotes covariance computation, and superscript *H* represents the complex conjugate transpose.

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As the source signal and noise signal are uncorrelated and mutually independent, the covariance
matrix C can be simplified as

289
$$\mathbf{C} = \mathbf{A}_{xy} \mathbf{R}_{s} \mathbf{A}_{xy}^{H} + \sigma^{2} \mathbf{I}, \qquad (20)$$

where $\mathbf{R}_s = E[s(t)gs(t)^H]$, and it signifies the covariance matrix of the source signal. σ^2 is noise power and **I** the covariance matrix of the noise signal. The covariance matrix **C** can be decomposed into two parts: namely a signal-related part and a noise-related part, as

293
$$\mathbf{C} = \mathbf{U}\boldsymbol{\Sigma}\mathbf{U}^{H} = \mathbf{U}_{S}\boldsymbol{\Sigma}\mathbf{U}_{S}^{H} + \mathbf{U}_{N}\boldsymbol{\Sigma}\mathbf{U}_{N}^{H}, \qquad (21)$$

where $\mathbf{U} = [\mu_1, \mu_2, \mathbf{L}, \mu_M]$, and the columns of \mathbf{U} are the singular vectors; $\boldsymbol{\Sigma}$ is a diagonal matrix with singular values arranged in a descending order of magnitudes. Considering that \mathbf{A}_{xy} is the steering vector at pixel (x, y) with the dimension of $M \times 1$ and $\mathbf{A}_{xy} \mathbf{R}_s \mathbf{A}_{xy}^{H}$ in Eq. (20) is decomposed as $\mathbf{U}_s \boldsymbol{\Sigma} \mathbf{U}_s^{H}$ in Eq. (21), $\mathbf{U}_s = [\mu_1]$ denoting the signal subspace spanned by the eigenvectors corresponding to the first largest eigenvalue; and $\mathbf{U}_N = [\mu_2, \mu_3, \mathbf{L}, \mu_M]$, representing the noise subspace spanned by the eigenvectors corresponding to the remaining M-1 eigenvalues.

301

302 Based on Eqs. (20) and (21), the following expression can be obtained after multiplying 303 covariance matrix **C** with the noise subspace \mathbf{U}_N

 $\mathbf{A}_{xy}\mathbf{R}_{s}\mathbf{A}_{xy}^{H}\mathbf{U}_{N}=\mathbf{0}.$ (22)

 $\mathbf{A}_{\mathbf{v}\mathbf{v}}^{H}\mathbf{U}_{N}=\mathbf{0}.$

(23)

305 As \mathbf{R}_s is a full rank matrix, Eq. (22) is further simplified as

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Equation (23) argues that the steering vector \mathbf{A}_{xy} at the position of damage is orthogonal with the noise subspace \mathbf{U}_N . This characteristic makes it possible for the Am-MUSIC to calculate the steering vector at each pixel across the entire inspection region and calibrate the degree of orthogonality between the steering vector and the noise subspace with the squared norm of vector $\mathbf{A}_{xy}^{\ H}\mathbf{U}_N$ as

312
$$\beta^2 = \left\| \mathbf{A}_{xy}^{\ H} \mathbf{U}_{N} \right\|^2 = \mathbf{A}_{xy}^{\ H} (\mathbf{U}_{N} \mathbf{U}_{N}^{\ H}) \mathbf{A}_{xy} .$$
(24)

Taking a reciprocal of the squared norm expression creates a peak in the spatial spectrum that corresponds to the damage location. Am-MUSIC algorithm defines the pixel value $(P_{\text{Am-MUSIC}}(x, y))$ within the inspection region as

316
$$P_{\text{Am-MUSIC}}(x, y) = \frac{1}{\mathbf{A}_{xy}^{H}(\mathbf{U}_{N}\mathbf{U}_{N}^{H})\mathbf{A}_{xy}}.$$
 (25)

Equation (24) yields a full spatial spectrum for the inspection region, in which $P_{\text{Am-MUSIC}}(x, y)$ culminates at the damage location. 319 In summary, the complete procedure of the proposed Am-MUSIC algorithm is flowcharted in a 320 nutshell in **Fig. 3**. Notably, the Am-MUSIC algorithm calculates the signal representation matrix 321 at each pixel throughout the entire inspection region, and it is therefore the number of scatterers 322 (i.e., the number of multiple damage sites) is no longer required, in contrast to a conventional 323 MUSIC algorithm in which the number of scatterers shall be predicted beforehand at a subjective 324 discretion of individuals. It is also noteworthy that the computational cost of Am-MUSIC does 325 not intend to increase compared with conventional MUSIC algorithms, regardless of the fact that 326 the calculation is performed at every signal pixel – that is because the signal representation matrix 327 can be formed cost-effectively and the steering factor in Eq. (18) can be calculated efficiently 328 compared with that of conventional MUSIC algorithm in Eq. (13). In addition, it is such a merit 329 of the Am-MUSIC algorithm that makes it possible to gauge the local region of interest (RoI) only 330 - the vicinity in the sample where damage may exist, rather than scanning the entire sample. Such 331 a merit remarkably lowers the computational cost and unburdens computing hardware when the 332 inspection region has of large dimensions.



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Fig. 3. Key steps of Am-MUSIC algorithm.

335 4. Numerical Validation

336 To validate the developed Am-MUSIC algorithm for damage imaging, numerical simulation is 337 implemented first. Consider a homogeneous, isotropic plate-like waveguide (density: p=2,700 kg/m³; Young modulus: E=71 GPa; Poisson's ratio v=0.33), measuring 500 mm \times 500 mm \times 2 338 339 mm. Atop the waveguide there is a sparse sensor network with only six PZT wafers, as illustrated 340 in Fig. 4(a). Each PZT wafer functions as either a wave transmitter or a wave receiver, leading to 341 15 transmitter-receiver pairs in the sensor network. For comparison against conventional MUSIC, 342 another seven PZT wafers are arranged in a linear array as sensors, in Fig. 4(b), along with an 343 additional PZT wafer as wave actuator placed at the position (250 mm, 400 mm). In all cases, a 344 3-cycle Hanning window tone burst with central frequency 200 kHz signal is selected as excitation signal to obtain S₀ wave mode, considering wave sensitivity and excitability. A total duration of 345 346 150 µs time length is analyzed for all numerical cases.





Fig. 4. Schematics of the plate waveguide in simulation (all dimensions in mm): (a) with a sparse sensor
 network for Am-MUSIC algorithm; and (b) with a linear array for conventional MUSIC algorithm.

350

351 Damage in the simulation is introduced to the waveguide by enforcing the material local stiffness

to be zero. Three damage sites, labeled as D1-D3, are simulated in the waveguide, with respective

353 positions highlighted in **Fig. 4** and summarized in **Table 1**. With these damage sites, three damage

354 cases (C-I – C-III) are created by including different damage sites, **Table 2**.

Table 1 Three damage sites in simulation

Damage site	Position	
	<i>x</i> [mm]	<i>y</i> [mm]
D1	200	200
D2	300	300
D3	230	230

357

Table 2 Three damage cases in simulation

Damage case	Damage site included
C-I	D1
C-II	D1, D2
C-III	D1, D3

358

Figure 5(a) displays the spatial spectrum obtained using the Am-MUSIC algorithm, for C-I – the 359 360 case with only a single damage site (D1), accurately pinpointing the damage location (200 mm, 361 200 mm). For comparison, the image constructed using the conventional MUSIC algorithm is 362 shown in Fig. 5(b), indicating the damage location at (197 mm, 208 mm), which represents an 363 error of (3 mm, 8 mm), in addition to an elongation artifact along the damage direction - a 364 common deficiency for conventional MUSIC algorithms as illustrated elsewhere [21-25, 36]. The 365 degree of such artifact depends on the signal-to-noise ratio and the point-spread function of the 366 phased array at the location of the scatterer [37].





Fig. 5. Spatial spectra for C-I obtained by (a) Am-MUSIC algorithm; and (b) conventional MUSIC
 algorithm.

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380 381

Figure 6(a) shows the spectrum for C-II obtained using the Am-MUSIC. Again, the identified results are observed to coincide exactly with actual damage sites, contrasting the spatial spectrum obtained using the conventional MUSIC algorithm in Fig. 6(b), in which two damage sites are predicted with notable error. In addition, the peak at D1 location is much stronger than that at D2 as noted in Fig. 6(b), contradicting the fact that D1 and D2 actually have the same degree of severity.





Fig. 6. Spatial spectra for C-II obtained by (a) Am-MUSIC algorithm; and (b) conventional MUSIC
 algorithm.

Provided that damage sites are in close proximity to each other – the case of C-III, the constructed spatial spectra using the Am-MUSIC method and conventional MUSIC method are compared in Fig. 7. In Fig. 7(a), the two damage sites are localized precisely, in good agreement with the actual positions; however, only one damage site is identified by the conventional MUSIC algorithm with remarkable artifacts, in Fig. 7(b), implying that the conventional MUSIC method may fail to detect multiple damage sites which are close one to another.





396

399

Fig. 7. Spatial spectra for C-III obtained by (a) Am-MUSIC algorithm; and (b) conventional MUSIC
 algorithm.

400 **5. Experimental Validation**

401 Subsequent to numerical simulation, effectiveness and accuracy of the Am-MUSIC-driven 402 anomaly imaging is validated experimentally. A 2 mm-thick aluminum plate (dimensions: 1000 mm \times 1000 mm \times 2 mm; density: ρ =2700 kg/m³; Young modulus: E=71 GPa; Poisson's ratio 403 404 v=0.33) is prepared. A sparse sensor network, consisting of eight PZT wafers (labelled as PZT-1, 405 PZT-2, ..., PZT-8), is surface-adhered on the plate, with the location of each wafer indicated in 406 Fig. 8(a). The experimental set-up is shown in Fig. 8(b). The excitation signal – a Hanning-407 window-modulated 5-cycle toneburst at a central frequency of 200 kHz - is generated with an arbitrary waveform generator (NI® PXI-5412) and amplified by a linear power amplifier (Ciprian® 408 409 US-TXP-3). The excitation signal is applied on each PZT wafer in turn to emit Lamb wave into 410 the plate. S₀ mode Lamb wave signals, each in 300 µs, are acquired with a digital oscilloscope (NI[®] PXI-5105) at a sampling rate of 60 MHz. 411



Damage Category	Damage Case	Damage
Single damage	E-I	A through-hole D1 at (400 mm, 400 mm)

		(Ø: 10 mm)
	E-II	A through-hole D2 at (300 mm, 700 mm)
		(Ø: 10 mm)
	E-III	A through-hole D3 at (740 mm, 250 mm)
		(Ø: 10 mm)
Double Damage	E-IV	Two through-holes D1 at (400 mm, 400
(long distance apart)		mm) and D4 at (600 mm, 600 mm)
		respectively (Ø: 10 mm)
	E-V	Two through-holes D1 at (400 mm, 400
		mm) and D2 at (300 mm, 700 mm)
		respectively (Ø: 10 mm)
Double Damage	E-VI	Two through-holes D1 at (400 mm, 400
(short distance apart)		mm) and D5 at (480 mm, 480 mm)
		respectively (Ø: 10 mm)

The spatial spectra constructed using the Am-MUSIC algorithm for all the six damage cases are presented in **Fig. 9**, in which all damage sites are accurately located with precise depiction of the damage shape. In particular, for E-VI in which two damage sites are close one to the other, the algorithm still warrants high resolution and distinguishes individual damage sites.





430 Fig. 9. Spatial spectra constructed using Am-MUSIC algorithm for damage case (a) E-I; (b) E-II; (c) E-431 III; (d) E-IV; (e) E-V; and (f) E-VI (red 'o': actual damage).

- 433 To take a step further, conventional MUSIC algorithm in junction with the use of a linear array is 434 recalled for comparison. Seven PZT wafers are configured in a linear array as receivers, in Fig. 435 10, along with an additional PZT wafer as a wave actuator placed at the position (500 mm, 800 436 mm). Four typical damage cases (E-I, E-III, E-IV, E-VI) are analysed. The spatial spectra 437 constructed using the conventional MUSIC algorithm are shown in Fig. 11, showing inferior 438 accuracy in damage localization and sizing; moreover, it fails to differentiate multiple damage 439 sites in E-VI that are close one from the other, and also fails to identify damage D3 in E-III that 440 is located in the blind zone for a conventional MUSIC algorithm.
- 441







Fig. 11. Spatial spectra constructed using conventional MUSIC algorithm for damage case (a) E-I; (b) E-III; (c) E-IV; and (d) E-VI (red 'o': actual damage).

447

448 **6. Concluding Remarks**

449 Aimed to circumvent some critical limitations of the conventional MUSIC algorithm-based 450 damage imaging, an ameliorated MUSIC algorithm is developed. In the Am-MUSIC algorithm, 451 the signal representation matrix at each pixel is manipulated by the excitation signal series, instead 452 of the scattered signal series, which enables the use of a sparse sensor network with arbitrarily 453 positioned transducers rather than a linear array featuring a dense configuration of transducing 454 elements with a uniform element pitch. By quantifying the orthogonal attributes between the 455 signal subspace and noise subspace inherent in the signal representation matrix, a full spatial 456 spectrum of the inspected sample can be generated, to visualize damage in the sample, irrespective 457 of the damage quantity. The effectiveness and accuracy of the Am-MUSIC algorithm are verified 458 in both simulation and experiment. Results show that compared with the conventional MUSIC 459 methods, the Am-MUSIC algorithm is capable of improving the detectability (in particular for 460 imaging of multiple damage sites that are close one to another) and eliminating blind zones. The 461 Am-MUSIC is conducive to expanding conventional MUSIC from phased array-facilitated 462 nondestructive evaluation to *in situ* health monitoring using built-in sparse sensor networks.

463

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471 **References**

[1] A.K.S. Jardine, D. Lin, D. Banjevic, A review on machinery diagnostics and prognostics
implementing condition-based maintenance, Mechanical Systems and Signal Processing, 20
(2006) 1483-1510.

[2] Z. Su, L. Ye, Identification of damage using Lamb waves: from fundamentals to applications,
 Springer Science & Business Media, 2009.

477 [3] W. Ostachowicz, P. Kudela, P. Malinowski, T. Wandowski, Damage localisation in plate-like

478 structures based on PZT sensors, Mechanical Systems and Signal Processing, 23 (2009) 1805-

479 1829.

- 480 [4] L. Zeng, J. Lin, Structural damage imaging approaches based on Lamb waves: A review, 2011
- 481 International Conference on Quality, Reliability, Risk, Maintenance, and Safety Engineering, 482 IEEE, 2011, pp. 986-993.
- 483 [5] P. Kudela, M. Radzienski, W. Ostachowicz, Z. Yang, Structural Health Monitoring system based on a concept of Lamb wave focusing by the piezoelectric array, Mechanical Systems and 484
- 485 Signal Processing, 108 (2018) 21-32.
- 486 [6] H. Gao, Y. Shi, J.L. Rose, Guided wave tomography on an aircraft wing with leave in place 487 sensors, AIP Conference Proceedings, American Institute of Physics, 2005, pp. 1788-1794.
- 488 [7] J.E. Michaels, Detection, localization and characterization of damage in plates with an in situ
- 489 array of spatially distributed ultrasonic sensors, Smart Materials and Structures, 17 (2008) 490 035035.
- 491 [8] L. Qiu, S. Yuan, X. Zhang, Y. Wang, A time reversal focusing based impact imaging method 492 and its evaluation on complex composite structures, Smart Materials and Structures, 20 (2011) 493 105014.
- 494 [9] C. Zhou, Z. Su, L. Cheng, Quantitative evaluation of orientation-specific damage using elastic
- 495 waves and probability-based diagnostic imaging, Mechanical Systems and Signal Processing, 25 496 (2011) 2135-2156.
- 497 [10] S.-J. Wei, X.-L. Zhang, J. Shi, G. Xiang, Sparse reconstruction for SAR imaging based on 498 compressed sensing, Progress In Electromagnetics Research, 109 (2010) 63-81.
- 499 [11] J.S. Hall, J.E. Michaels, Minimum variance ultrasonic imaging applied to an in situ sparse 500 guided wave array, IEEE transactions on ultrasonics, ferroelectrics, and frequency control, 57
- 501 (2010) 2311-2323.
- 502 [12] M. Viberg, B. Ottersten, Sensor array processing based on subspace fitting, IEEE 503 Transactions on signal processing, 39 (1991) 1110-1121.
- [13] A.J. Hartemink, D.K. Gifford, T.S. Jaakkola, R.A. Young, Maximum-likelihood estimation 504
- 505 of optimal scaling factors for expression array normalization, Microarrays: Optical technologies 506 and informatics, International Society for Optics and Photonics, 2001, pp. 132-140.
- 507 [14] R. Schmidt, Multiple emitter location and signal parameter estimation, IEEE transactions on 508 antennas and propagation, 34 (1986) 276-280.
- 509 [15] R.O. Schmidt, A signal subspace approach to multiple emitter location and spectral 510 estimation, (1982).
- 511 [16] R. Compton, Two-dimensional imaging of radar targets with the MUSIC algorithm, 1987.
- 512 [17] T. Iwata, Y. Goto, H. Susaki, Application of the multiple signal classification (MUSIC)
- 513 method for one-pulse burst-echo Doppler sonar data, Measurement Science and Technology, 12 514 (2001) 2178.
- 515 [18] P. Goldstein, R.J. Archuleta, Array analysis of seismic signals, Geophysical Research Letters, 516 14 (1987) 13-16.
- 517 [19] J.C. Mosher, S. Baillet, R.M. Leahy, EEG source localization and imaging using multiple 518 signal classification approaches, Journal of Clinical Neurophysiology, 16 (1999) 225-238.
- 519 [20] M. Engholm, T. Stepinski, Direction of arrival estimation of Lamb waves using circular 520 arrays, Structural Health Monitoring, 10 (2011) 467-480.
- 521 [21] H. Yang, T.J. Shin, S. Lee, Source location in plates based on the multiple sensors array 522 method and wavelet analysis, Journal of Mechanical Science and Technology, 28 (2014) 1-8.
- 523 [22] H. Yang, Y.J. Lee, S.K. Lee, Impact source localization in plate utilizing multiple signal 524 classification, Proceedings of the Institution of Mechanical Engineers, Part C: Journal of
- 525 Mechanical Engineering Science, 227 (2013) 703-713.
- 526 [23] Y. Zhong, S. Yuan, L. Qiu, Multiple damage detection on aircraft composite structures using 527 near-field MUSIC algorithm, Sensors and Actuators A: Physical, 214 (2014) 234-244.
- 528 [24] Y. Zhong, S. Yuan, L. Qiu, An improved two-dimensional multiple signal classification approach for impact localization on a composite structure, Structural Health Monitoring, 14
- 529
- 530 (2015) 385-401.

- 531 [25] S. Yuan, Q. Bao, L. Qiu, Y. Zhong, A single frequency component-based re-estimated
- 532 MUSIC algorithm for impact localization on complex composite structures, Smart Materials and 533 Structures, 24 (2015) 105021.
- 534 [26] H. Zuo, Z. Yang, C. Xu, S. Tian, X. Chen, Damage identification for plate-like structures
- using ultrasonic guided wave based on improved MUSIC method, Composite Structures, 203(2018) 164-171.
- 537 [27] Q. Bao, S. Yuan, F. Guo, L. Qiu, Transmitter beamforming and weighted image fusion-based
- multiple signal classification algorithm for corrosion monitoring, Structural Health Monitoring,
 18 (2019) 621-634.
- 540 [28] Q. Bao, S. Yuan, Y. Wang, L. Qiu, Anisotropy compensated MUSIC algorithm based 541 composite structure damage imaging method, Composite Structures, 214 (2019) 293-303.
- 542 [29] V. Giurgiutiu, Structural health monitoring: with piezoelectric wafer active sensors, Elsevier,
- 543 2007.
- 544 [30] S. Sundararaman, D.E. Adams, E.J. Rigas, Structural damage identification in homogeneous
- and heterogeneous structures using beamforming, Structural Health Monitoring, 4 (2005) 171-190.
- 547 [31] A. Purekar, D. Pines, S. Sundararaman, D. Adams, Directional piezoelectric phased array 548 filters for detecting damage in isotropic plates, Smart Materials and Structures, 13 (2004) 838.
- 549 [32] Y. Zhong, S. Yuan, L. Qiu, Multi-impact source localisation on aircraft composite structure
- 550 using uniform linear PZT sensors array, Structure and Infrastructure Engineering, 11 (2015) 310-
- 551 320.
- 552 [33] Y. Wang, K. Ho, Unified near-field and far-field localization for AOA and hybrid AOA-553 TDOA positionings, IEEE Transactions on Wireless Communications, 17 (2017) 1242-1254.
- 554 [34] E. Grosicki, K. Abed-Meraim, Y. Hua, A weighted linear prediction method for near-field 555 source localization, IEEE Transactions on Signal Processing, 53 (2005) 3651-3660.
- 556 [35] A. Swindlehurst, T. Kailath, Passive direction-of-arrival and range estimation for near-field 557 sources, IEEE Spec. Est. and Mod. Workshop, Citeseer, 1988.
- 558 [36] Q. Bao, S. Yuan, F. Guo, A new synthesis aperture-MUSIC algorithm for damage diagnosis 559 on complex aircraft structures, Mechanical Systems and Signal Processing, 136 (2020) 106491.
- 560 [37] Y. Labyed, L. Huang, Ultrasound time-reversal MUSIC imaging with diffraction and
- attenuation compensation, IEEE transactions on ultrasonics, ferroelectrics, and frequency control,
- 562 59 (2012) 2186-2200.
- 563